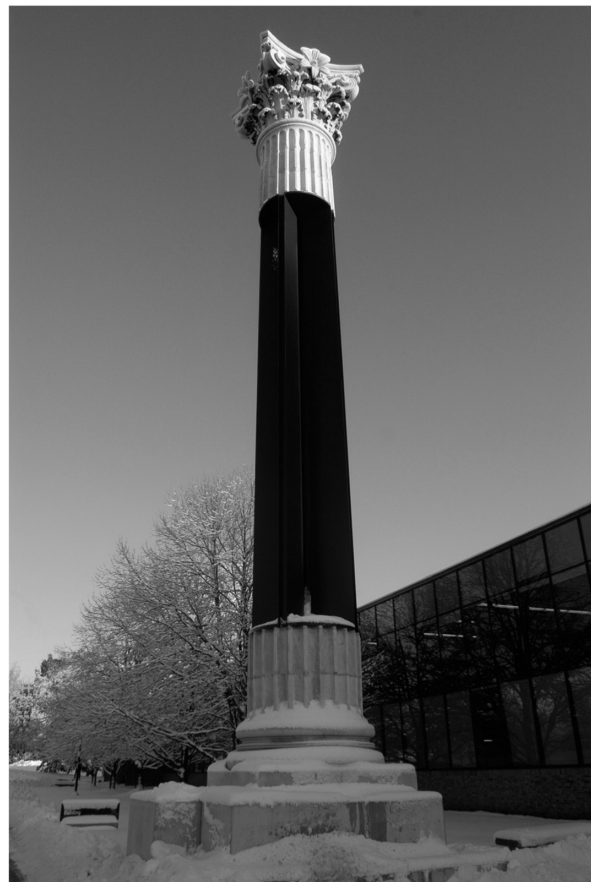


Steel Column Analysis

- Failure Modes
- Effects of Slenderness
- Stress Analysis of Steel Columns



Leonhard Euler (1707 – 1783)

Euler Buckling (elastic buckling)

$$\underline{P_{cr}} = \frac{\pi^2 AE}{\left(\frac{KL}{r}\right)^2} = \frac{\pi^2 IE}{(KL)^2}$$

$$r = \sqrt{\frac{I}{A}}$$

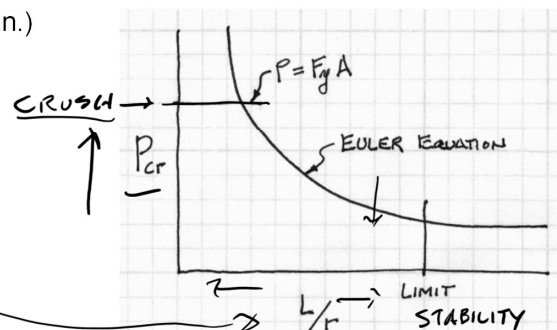
$$\underline{I = Ar^2}$$

- A = Cross sectional area (in²)
- E = Modulus of elasticity of the material (lb/in²)
- K = Stiffness (curvature mode) factor
- L = Column length between pinned ends (in.)
- r = radius of gyration (in.)



portrait by Emanuel Handmann, 1753

$$\underline{f_{cr}} = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} \leq F_{cr}$$



Analysis of Steel Columns

Conditions of an Ideal Column

- initially straight ✓
- axially loaded
- uniform stress (no residual stress)
- uniform material (no holes)
- no transverse load
- pinned (or defined) end conditions



Analysis of Steel Columns

Short columns

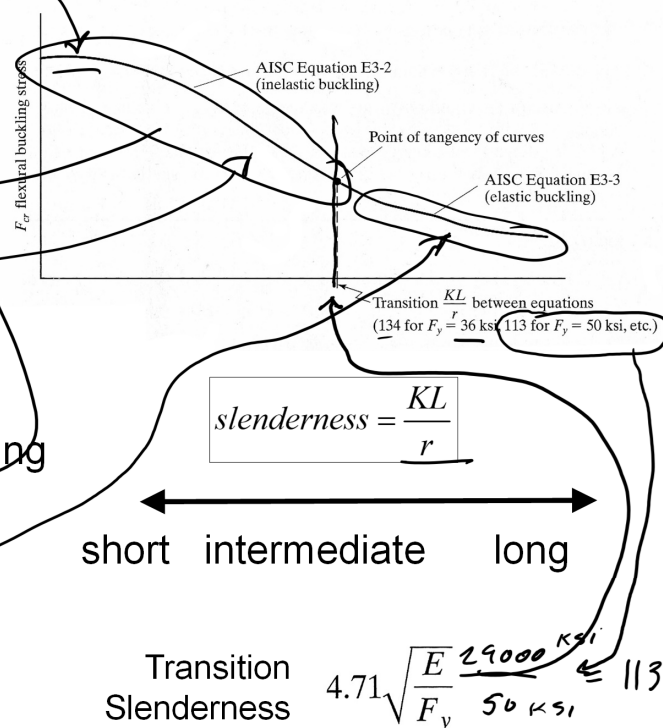
Fail by material crushing
Plastic behavior

Intermediate columns

Crush partially and then buckle
Inelastic behavior
Local buckling – flange or web
Flexural torsional buckling - twisting

Long columns

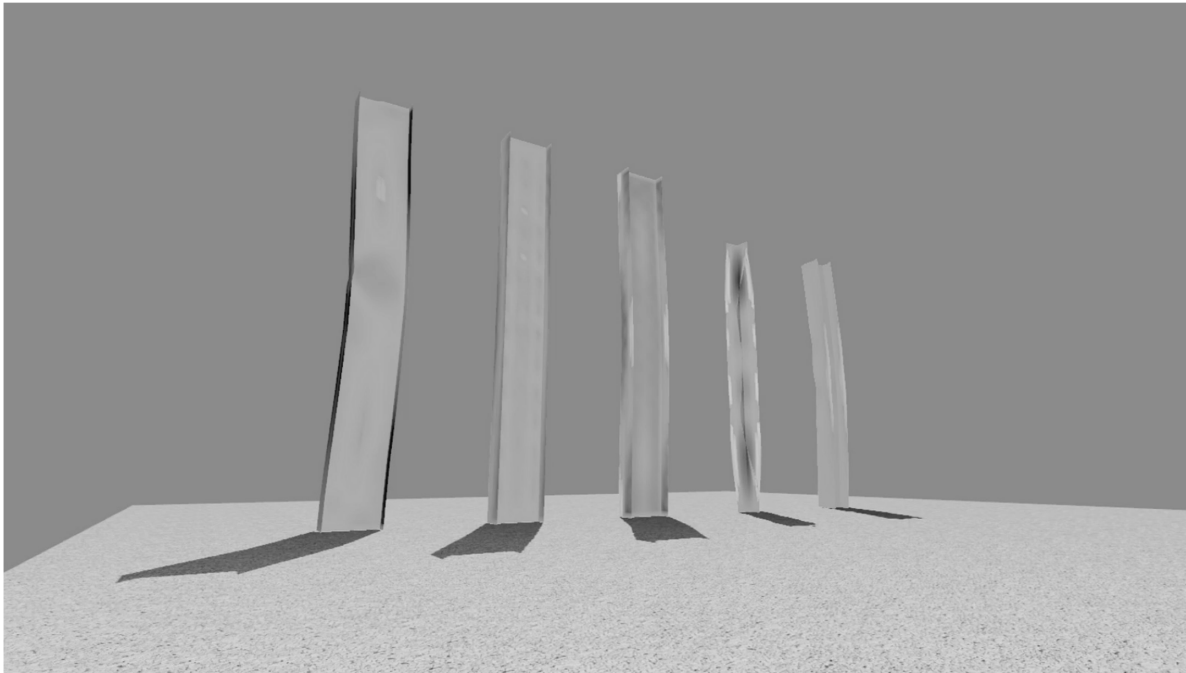
Fail in Euler buckling
Elastic behavior



Failure Modes

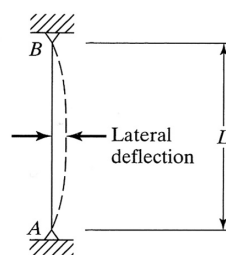
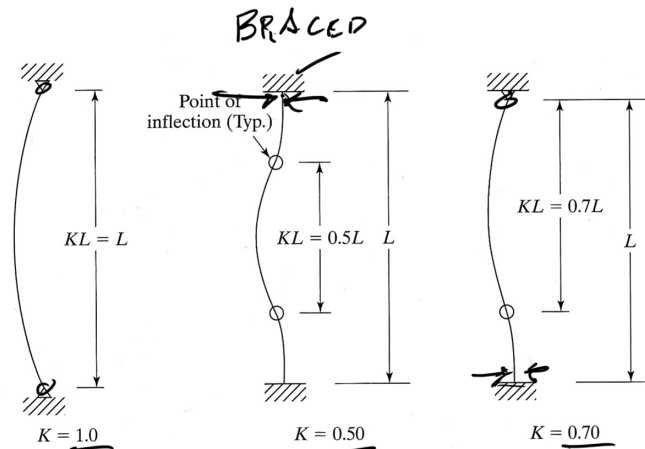
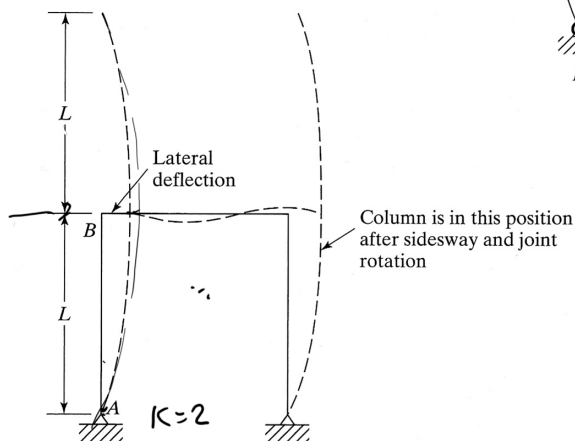
- Column 1: Strong axis flexural buckling
- Column 2: Web local buckling
- Column 3: Weak axis flexural buckling
- Column 4: Torsional buckling
- Column 5: Flange local buckling

"Dancing Columns"
Sherif El-Tawil



Analysis of Steel Columns

Estimate of
effective length factor, K



Determining K factors by Alignment Charts

Sidesway uninhibited

Sidesway Inhibited:
Braced frame
 $1.0 > K > 0.5$

Sidesway Uninhibited:
Un-braced frame
 $\text{unstable} > K > 1.0$

More Pinned:
If I_c/L_c is large
and I_g/L_g is small
The connection is more pinned
and in this case unstable

More Fixed:
If I_c/L_c is small
and I_g/L_g is large
The connection is more fixed

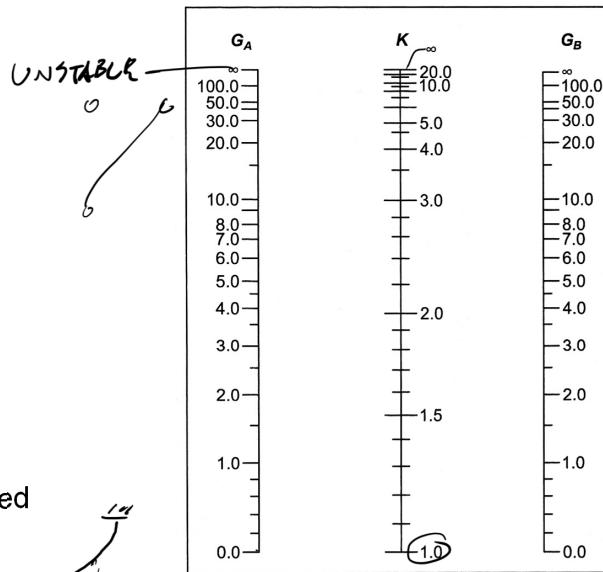


Fig. C-A-7.2. Alignment chart—sidesway uninhibited (moment frame).

$$G = \frac{\sum \left(\frac{EI}{L} \right)_{\text{column}}}{\sum \left(\frac{EI}{L} \right)_{\text{beam}}}$$

Analysis of Steel Columns - LRFD

Euler equation:

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r} \right)^2}$$

Short & Intermediate Columns:

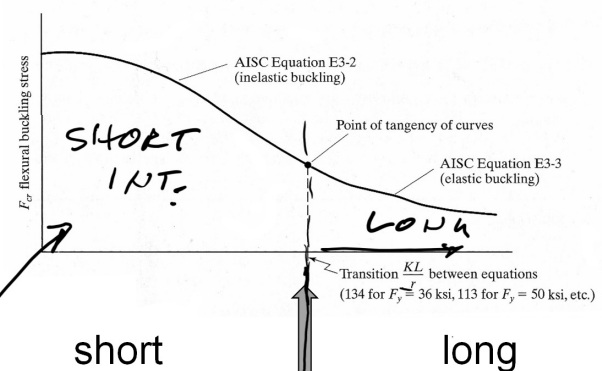
$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}} \right] F_y$$

Equation E3-2

Long Columns:

$$F_{cr} = 0.877(F_e)$$

Equation E3-3



Transition
Slenderness

$$4.71 \sqrt{\frac{E}{F_y}}$$

$$P_n = F_{cr} A_g$$

$$\phi_c P_n = \phi_c F_{cr} A_g > \phi_u$$

($\phi_c = 0.90$)

Analysis of Steel Columns

pass / fail by LRFD

Data:

- Column – size, length
- Support conditions
- Material properties – F_y
- Factored load – P_u

Required:

- $P_u \leq \phi P_n$ (pass)

1. Calculate slenderness ratios: L_c/r_x and L_c/r_y ($L_c = KL$)
The largest ratio governs.
2. Check slenderness ratio against upper limit of 200 (recommended)
3. Calculate transition slenderness $4.71\sqrt{E/F_y}$ and determine column type (short or long)
4. Calculate F_{cr} based on slenderness
5. Determine ϕP_n and compare to P_u
 $P_n = F_{cr} A_g$ $\phi = 0.9$
6. If $P_u \leq \phi P_n$, then OK

$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}} \right] F_y \quad \text{Short}$$

$$F_{cr} = 0.877 F_e \quad \text{Long}$$



Example - Analysis of Steel Columns

pass / fail by ASD

Data:

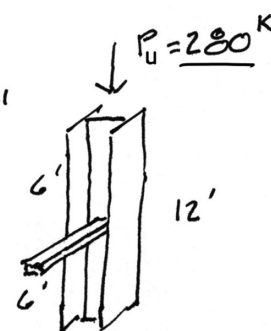
- Column – size, length
- Support conditions
- Material properties – F_y
- Factored Load – P_u

Required:

- $P_u \leq \phi P_n$ (pass)

DATA :

$W 8 \times 35$
 $r_x = 3.51''$
 $r_y = 2.03''$
 $A = 10.3 \text{ in}^2$
 $l_x = 12'$ $l_y = 6'$
 $K_x = K_y = 1.0$



1. Calculate slenderness ratios: L_c/r_x and L_c/r_y ($L_c = KL$)
The largest ratio governs.

The diagram illustrates the standard dimensions of a W-shape cross-section. It shows a central web with thickness t_w and two flanges with thickness t_f . The total depth is d , and the total flange width is b_f . The distance between the centerlines of the flanges is k . The diagram also indicates the location of the centroidal axes (X-X and Y-Y) and the corresponding moments of inertia I_x and I_y .

Table 1-1 (continued)
W-Shapes
Dimensions

Shape	Area, A	Depth, d	Web		Flange		Distance				Workable Gage			
			Thickness, t _w	t _w / 2	Width, b _f	Thickness, t _f	k	k _{des}	k _{det}	T				
	in. ²	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.			
W8x67	19.7	9.00	9	0.570	5/16	8.28	8 1/4	0.935	15/16	1.33	1 5/8	15/16	5 3/4	5 1/2
x58	17.1	8.75	8 3/4	0.510	1/2	7 1/4	8 1/4	0.810	13/16	1.20	1 1/2	7/8		
x48	14.1	8.50	8 1/2	0.400	3/8	7 3/8	8 1/4	0.685	11/16	1.08	1 3/8	13/16		
x40	11.7	8.25	8 1/4	0.360	3/8	7 1/8	8 1/4	0.560	9/16	0.954	1 1/4	13/16		
x35	10.3	8.12	8 1/8	0.310	5/16	7 1/8	8 1/4	0.495	1/2	0.889	13/16	13/16		
x31	9.13	8.00	8	0.285	5/16	7 1/8	8 1/4	0.435	7/16	0.829	1 1/8	3/4		

Table 1-1 (continued)
W-Shapes
Properties

Nominal Wt.	Compact Section Criteria		Axis X-X				Axis Y-Y				r_ts	h_o	Torsional Properties	
	b_f/2t_f	h/t_w	I	S	r	Z	I	S	r	Z			J	C_w
lb/ft	2t_f	t_w	in. ⁴	in. ³	in.	in. ³	in. ⁴	in. ³	in.	in. ³	in.	in.	in. ⁴	in. ⁶
67	4.43	11.1	272	60.4	3.72	70.1	88.6	21.4	2.12	32.7	2.43	8.07	5.05	1440
58	5.07	12.4	228	52.0	3.65	59.8	75.1	18.3	2.10	27.9	2.39	7.94	3.33	1180
48	5.92	15.9	184	43.2	3.61	49.0	60.9	15.0	2.08	22.9	2.35	7.82	1.96	931
40	7.21	17.6	146	35.5	3.53	39.8	49.1	12.2	2.04	18.5	2.31	7.69	1.12	726
35	8.10	20.5	127	31.2	3.47	34.7	42.6	10.6	2.03	16.1	2.28	7.63	0.769	619
31	9.19	22.3	110	27.5	3.47	30.4	37.1	9.27	2.02	14.1	2.26	7.57	0.536	530

Example - Analysis of Steel Columns

pass / fail by ASD

Data:

- Column – size, length
- Support conditions
- Material properties – F_y
- Factored Load – P_u

Required:

- $P_u \leq \phi P_n$ (pass)

- Calculate slenderness ratios.
The largest ratio governs.
- Check slenderness ratio against upper limit of 200 (recommended)

DATA :

$W 8 \times 35$ A-36
 $r_x = 3.51''$ $F_y = 36 \text{ ksi}$
 $r_y = 2.03''$
 $A = 10.3 \text{ in}^2$

$l_x = 12'$ $l_y = 6'$
 $K_x = K_y = 1.0$

X-X AXIS

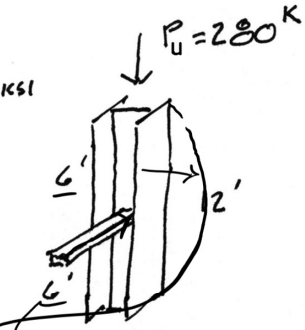
$$\frac{K_x l_x}{r_x} = \frac{144}{3.51''}$$

41.03 < 200 ✓

Y-Y AXIS

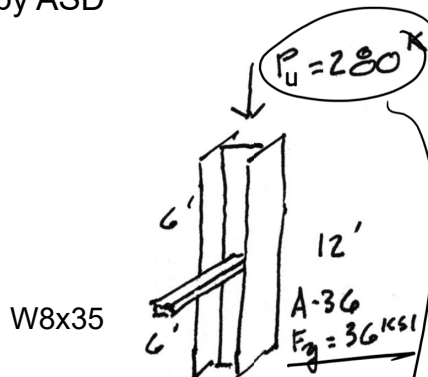
$$\frac{K_y l_y}{r_y} = \frac{72''}{2.03''}$$

35.47



Example - Analysis of Steel Columns

pass / fail by ASD



$$4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{29000}{36}} = 134$$

41 < 134 ∴ SHORT

Euler Equation

$$F_e = \frac{\pi^2 E}{\left(\frac{K L}{r}\right)^2} = \frac{\pi^2 29000 \text{ ksi}}{41^2} = 170.2 \text{ ksi}$$

Short Column Equation

$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}} \right] F_y = 0.9153 (36) = 32.95 \text{ ksi}$$

Column Strength

$$P_n = F_{cr} A_g = 32.95 \text{ ksi} \times 10.3 \text{ in}^2 = 339.39 \text{ K}$$

$$\phi P_n = 0.9 P_n = 0.9 (339.39) = 305.4 \text{ K}$$

$P_u = 280 \text{ K} < 305.4 \text{ K} = \phi P_n$ ✓ OK
 LOAD STRENGTH

- Calculate transition slenderness $4.71 \sqrt{E/F_y}$ and determine column type (short or long)
- Calculate F_{cr} based on slenderness
- Determine ϕP_n and compare to P_u
- If $P_u \leq \phi P_n$, then OK

Analysis of Steel Columns capacity by LRFD

Data:

- Column – size, length
- Support conditions
- Material properties – F_y

Required:

- Max load capacity

1. Calculate slenderness ratios.
The largest ratio governs. ✓

2. Check slenderness ratio against upper limit of 200 (recommended) ✓

3. Calculate transition slenderness $4.71\sqrt{E/F_y}$ and determine column type (short or long)

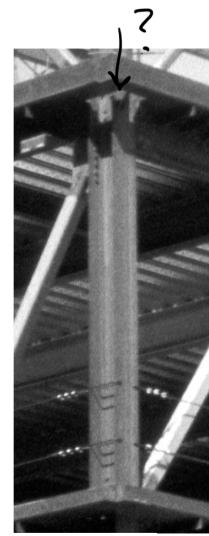
$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}} \right] F_y \quad \text{Short}$$

4. Calculate F_{cr} based on slenderness

$$F_{cr} = 0.877 F_e \quad \text{Long}$$

5. Determine ϕP_n and Compute allowable capacity:

$$P_n = F_{cr} A_g \quad \phi P_u = \phi P_n$$

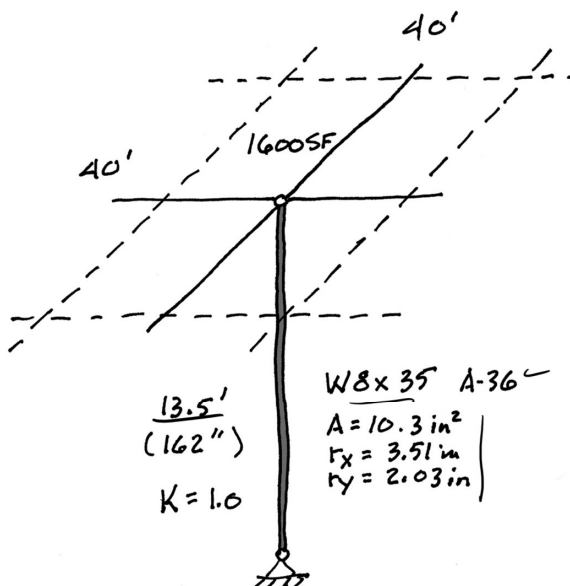


Capacity Example 1

Free standing column

Third floor studio space

Supports roof load = 20 psf DL + SL
snow $\approx 15 \text{ lbs / FT depth}$



Capacity Example 1

1. Calculate slenderness ratios.
The largest ratio governs.
2. Check slenderness ratio against upper limit of 200 (recommended)
3. Calculate transition slenderness $4.71\sqrt{E/F_y}$ and determine column type (short or long)
4. Calculate F_{cr} based on slenderness

y-y Axis (CONTROLS)
Pinned

$$\frac{K_y L_y}{r_y} = \frac{(1)(162')}{2.03'} = 79.8 < 200 \checkmark$$

$$4.71\sqrt{\frac{E}{F_y}} = 4.71\sqrt{\frac{29000}{36}} = 134$$

$$79.8 < 134 \therefore \text{SHORT}$$

Euler Buckling

$$F_e = \frac{\pi^2 E}{(K L/r)^2} = \frac{\pi^2 29000}{79.8^2} = 44.94 \text{ ksi}$$

Short Column Equation

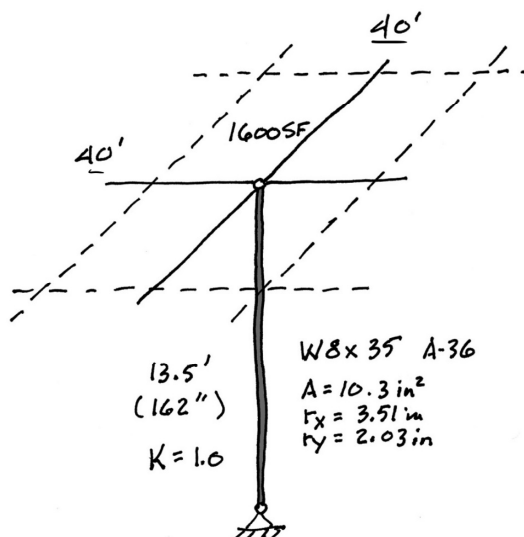
$$F_{cr} = \left[0.658^{\frac{F_e}{F_y}} \right] F_y = \left[0.7151 \right] 36 = 25.74 \text{ ksi}$$

Capacity Example 1

5. Determine ϕP_n and Compute allowable capacity: $P_u = \phi P_n$

DL = 20 psf

20 psf (1600 sf) = 32k on column



Column nominal strength

$$P_n = F_{cr} A_g = 25.74 \text{ ksi} \cdot 10.3 \text{ in}^2 = 265.1 \text{ k}$$

$$\phi P_n = 0.9(265) = 238.6 \text{ k} = P_u$$

CAPACITY

Load capacity

$$P_u = 1.2(32) + 1.6(5L) = 238.6 \text{ k}$$

SL = 125.1'

$$\text{For } A_t = 40 \times 40 = 1600 \text{ SF}$$

$$SL = \frac{125100^*}{1600 \text{ SF}} = 78.2 \text{ PSF}$$

$$78.2 \text{ lbs} / 15 \text{ lbs/ft} = 5.21 \text{ ft}$$

Capacity Example 2

long column – using equations

Find the capacity for the
25 ft. column shown.

$$r_x = 3.51 \text{ in.}$$

$$r_y = 2.03 \text{ in.}$$

Table G1 Buckling Length Coefficients, K_e

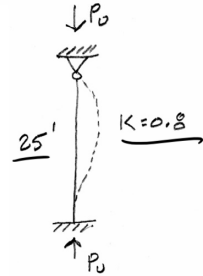
Buckling modes						
Theoretical K_e value	0.5	0.7	1.0	1.0	2.0	2.0
Recommended design K_e when ideal conditions approximated	0.65	0.80	1.2	1.0	2.10	2.4
End condition code		Rotation fixed, translation fixed				
		Rotation free, translation fixed				
		Rotation fixed, translation free				
		Rotation free, translation free				

$$W8 \times 35$$

$$F_y = 50 \text{ ksi}$$

$$E = 29,000 \text{ ksi}$$

$$L = 25' \text{ (No Bracing)}$$



Slenderness y-y

$$\frac{KL_y}{r_y} = \frac{0.8(25')}{2.03} = 118.2$$

$$4.71\sqrt{\frac{E}{F_y}} = 113 < 118.2 \therefore \text{LONG}$$

Euler Buckling

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = \frac{\pi^2 29,000}{118.2^2} = 20.47 \text{ ksi}$$

Long Column Equation

$$F_{cr} = 0.877(20.47) = 17.95 \text{ ksi}$$

Column strength

$$\phi P_n = \phi F_{cr} A_g = 0.9(17.95)(10.3) = 166.4 \text{ k}$$

Capacity Example 2

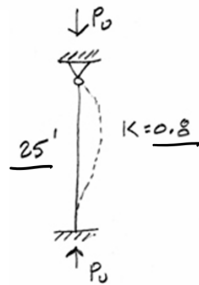
long column – using table

$$W8 \times 35$$

$$F_y = 50 \text{ ksi}$$

$$E = 29,000 \text{ ksi}$$

$$L = 25' \text{ (No Bracing)}$$



r_y CONTROLS

$$L_c = KL = 0.8(25') = 20'$$

Table 4-1a (continued) Available Strength in Axial Compression, kips													
W-Shapes													
LRFD													
Shape	67		58		48		40		35		31		
lb/ft													
Design	P_n/Ω_c	$\phi_c P_n$	P_n/Ω_c	$\phi_c P_n$	P_n/Ω_c	$\phi_c P_n$	P_n/Ω_c	$\phi_c P_n$	P_n/Ω_c	$\phi_c P_n$	P_n/Ω_c	$\phi_c P_n$	
	ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD	
0	590	886	512	769	422	634	350	526	308	463	273	411	
6	542	815	470	706	387	581	320	481	281	423	249	374	
7	526	790	455	685	375	563	309	465	272	409	241	362	
8	508	763	439	660	361	543	298	448	262	394	232	346	
9	488	733	422	634	347	521	285	429	251	377	222	333	
10	467	701	403	606	331	497	272	409	239	359	211	317	
11	444	668	384	576	314	473	258	388	226	340	200	301	
12	421	633	363	546	297	447	243	366	213	321	189	283	
13	397	597	342	514	280	421	228	343	200	301	177	266	
14	373	560	321	482	262	394	213	321	187	281	165	248	
15	348	523	299	450	244	367	198	298	174	261	153	230	
16	324	487	278	418	226	340	183	275	160	241	141	212	
17	300	450	257	386	209	314	169	253	147	221	130	195	
18	276	415	236	355	192	288	154	232	135	203	118	178	
19	253	381	216	325	175	264	141	211	123	183	106	162	
20	231	347	197	296	159	239	127	191	111	166	97.2	146	
22	191	287	163	244	132	198	105	158	91.5	138	80.3	121	
24	160	241	137	205	111	166	88.2	133	76.9	116	67.5	101	
26	137	205	116	175	94.2	142	75.2	113	65.5	98.5	57.5	86.5	
28	118	177	100	151	81.2	122	64.8	97.4	56.5	84.9	49.6	74.5	
30	103	154	87.5	131	70.7	106	56.5	84.9	49.2	74.0	43.2	64.9	
32	90.3	136	76.9	116	62.2	93.5	49.6	74.6	43.3	65.0	38.0	57.1	
34	79.9	120	68.1	102	55.1	82.8	44.0	66.1					
Properties													
P_{n0} , kips	126	190	102	153	72.0	108	57.2	85.9	45.9	68.9	39.4	59.1	
P_{n1} , kip/in.	19.0	28.5	17.0	25.5	13.3	20.0	12.0	18.0	10.3	15.5	9.50	14.3	
P_{n2} , kips	507	761	363	546	174	262	127	192	81.1	122	63.0	94.7	
P_{n3} , kips	164	246	123	185	87.8	132	68.7	88.2	45.9	68.9	35.4	53.2	
L_p , ft	7.49		7.42		7.35		7.21		7.17		7.18		
L_r , ft	47.6		41.6		35.2		29.9		27.0		24.8		
A_g , in. ²	19.7		17.1		14.1		11.7		10.3		9.13		
I_y , in. ⁴	272		228		184		146		127		110		
I_x , in. ⁴	88.6		75.1		60.9		49.1		42.6		37.1		
r_y , in.	2.12		2.10		2.08		2.04		2.04		2.02		
r_x , in.	1.75		1.74		1.74		1.73		1.78		1.72		
$P_{n0}L^2/10^4$, k-in. ²	7790		6530		5270		4190		3630		3150		
$P_{n1}L^2/10^4$, k-in. ²	2540		2150		1740		1410		1220		1060		
ASD													
LRFD													
Note: Heavy line indicates L_c/r_y equal to or greater than 200.													
$\Omega_c = 1.67$ $\phi_c = 0.90$													