

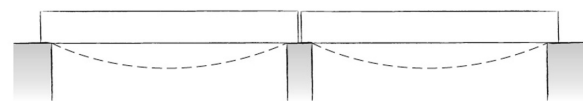
Gerber Beams

- Continuity in Beams
- Gerber Beams
- Optimization

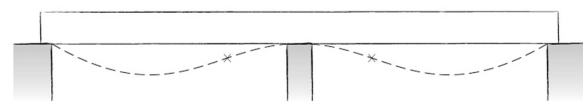


Continuous Beams

- Continuous over one or more supports
 - Most common in monolithic concrete
 - Steel: continuous or with moment connections
 - Wood: as continuous beams, e.g. long Glulam spans
- Statically indeterminate
 - Cannot be solved by the three equations of statics alone
 - Internal forces (shear & moment) as well as reactions are affected by movement or settlement of the supports



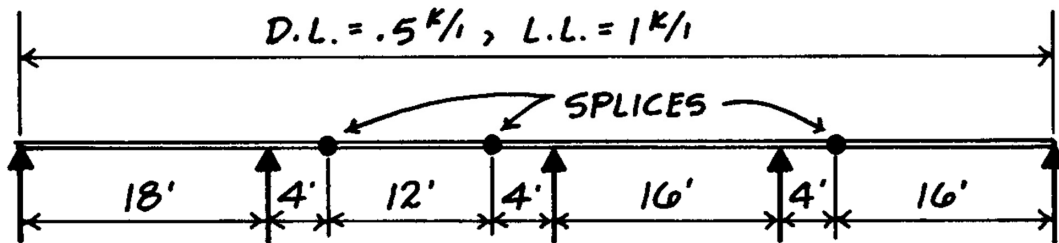
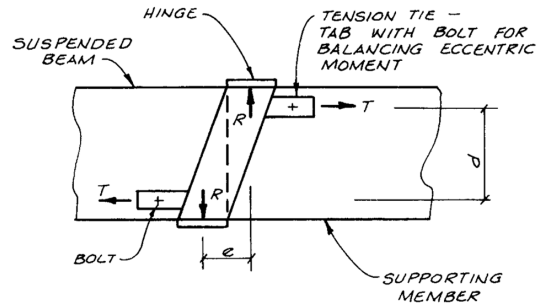
two spans - simply supported



two spans - continuous

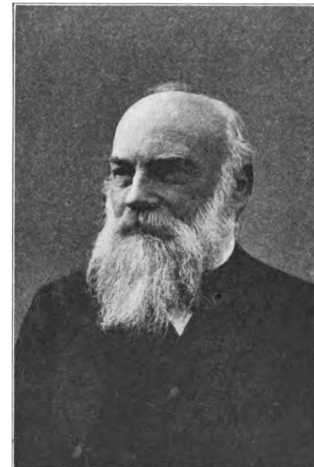
Splice or Hinge

- Can add one hinge for each redundant reaction
- Reduces length for transport
- Moment = 0 at hinge
- Can be used to balance – and + moments for optimization

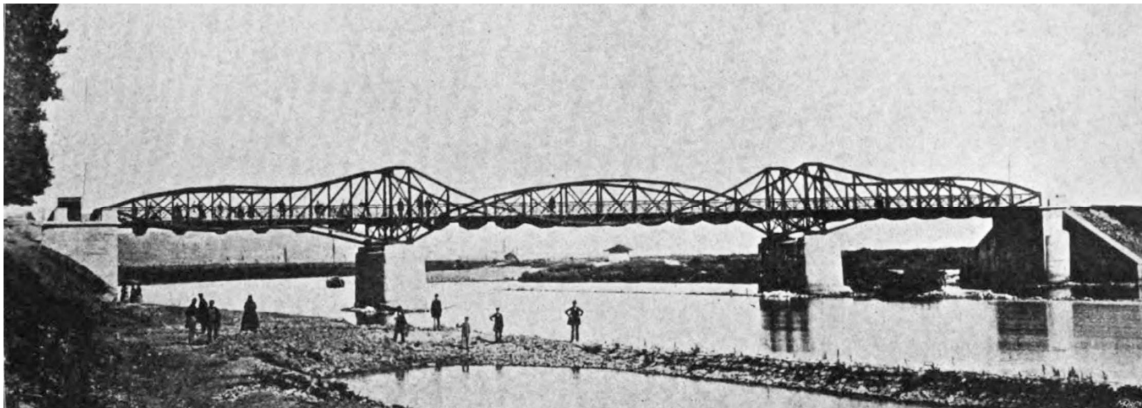


Gottfried Heinrich Gerber (1832-1912)

Developed a cantilever bridge spanning system used in many bridges worldwide. The system became known as the “Gerber Beam” and uses cantilever segments to support a simple span.



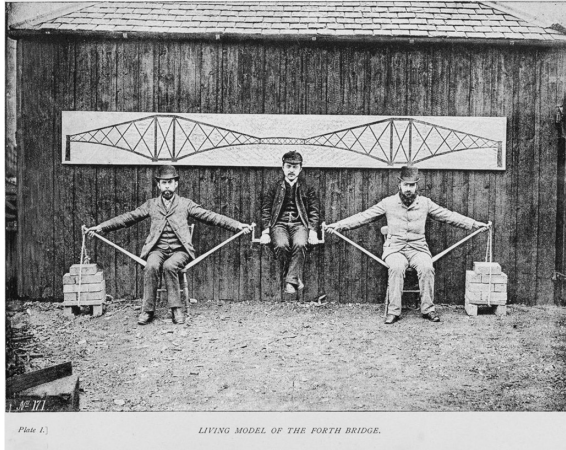
Haßfurter Brücke, 1864. Span of 38 m over the Main River.



Examples of the Gerber system

Firth of Forth Bridge, 1890

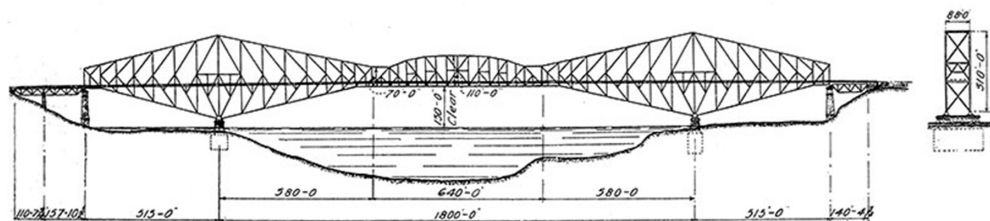
- total length 8094 ft.
- central span 1700 ft.
- Design Fowler & Baker
- Construction 1882 - 1889



Static modeling of the Firth of Forth Bridge
by Fowler & Baker



Quebec Bridge Final Completion 1917



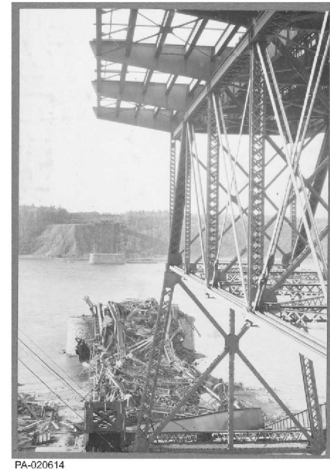
Final successful completion 1917

Quebec Bridge failure – 1907 and 1916

Compression members that failed in 1907

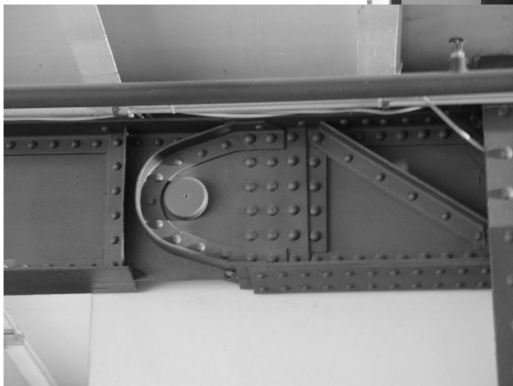


1916 hoisting failure



1907 failure due to miscalculation of the steel strength and dead load.

Gerber system in building frames



Speicherstadt Hamburg Kaffeerösterei
1888

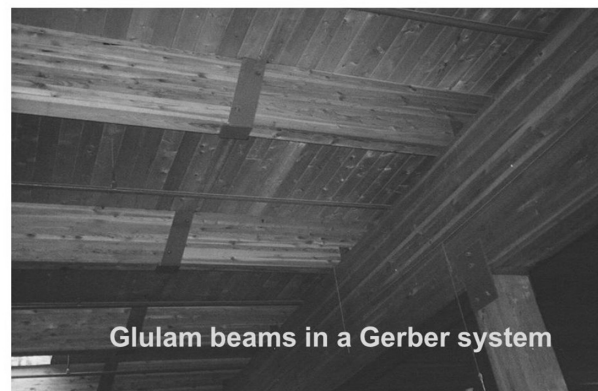
Gerber Beams in Detroit



Example Gerber Beams

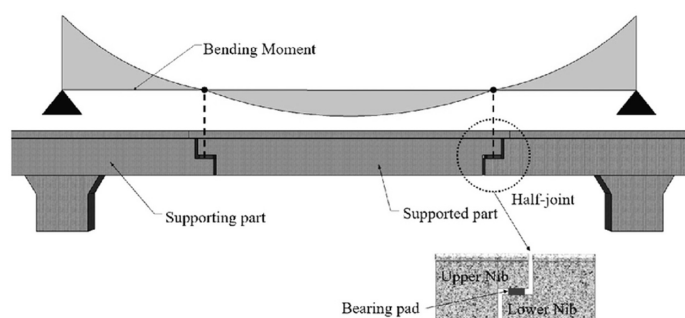


Steel

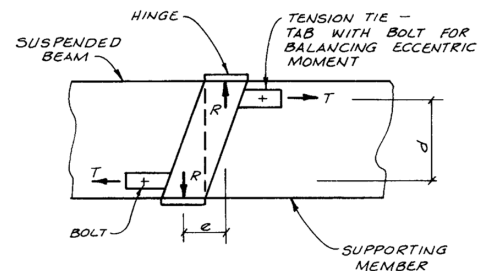


Glulam beams in a Gerber system

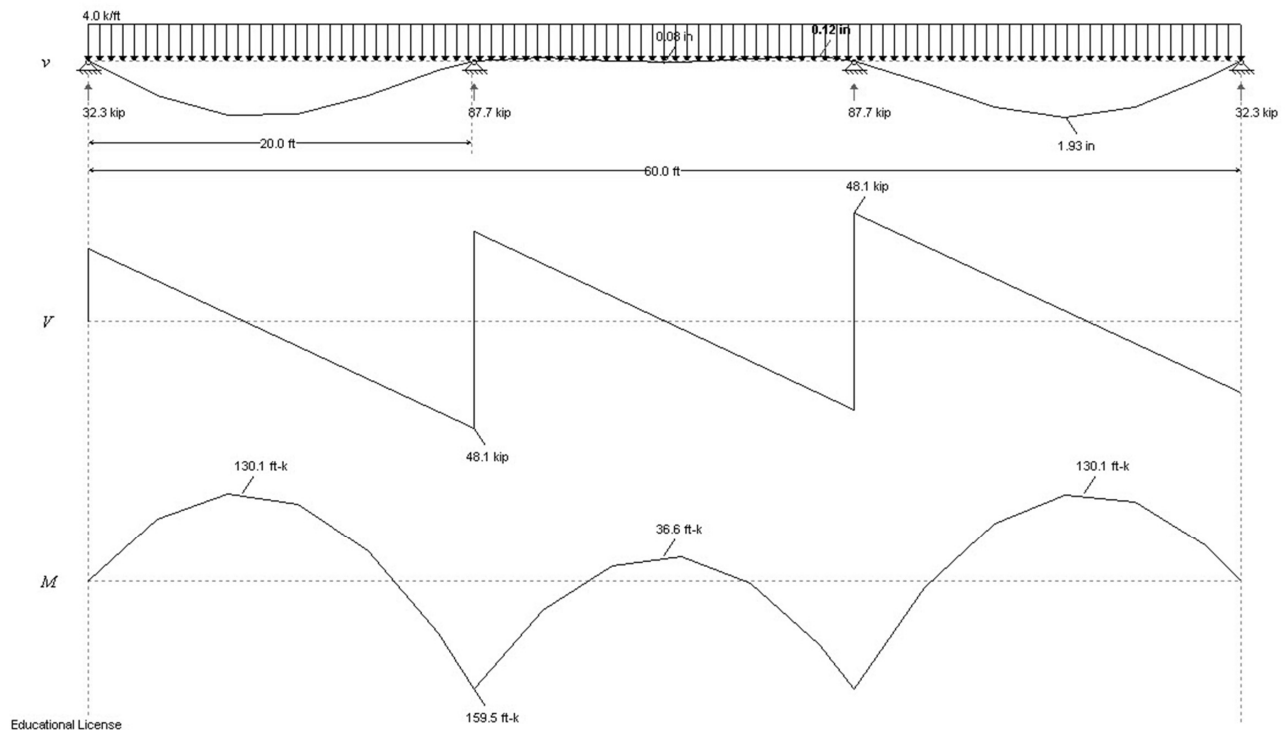
Concrete



Wood

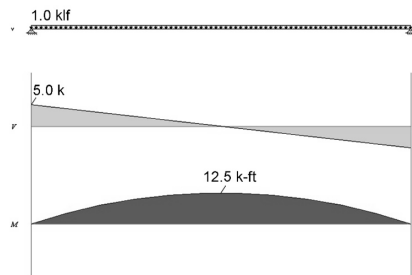


Moment control in beams

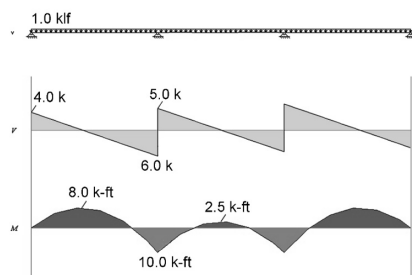


Moment control in beams

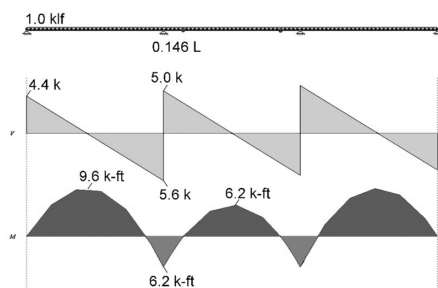
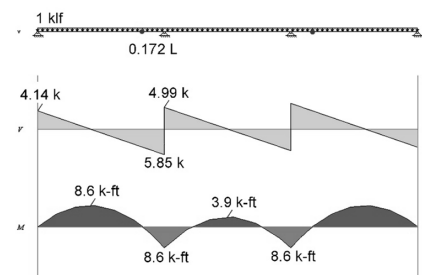
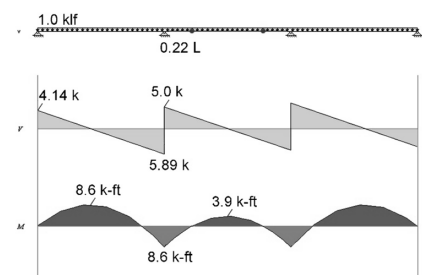
Spans = 10 ft



simple span



three spans – without hinges



three spans – with hinges

Example Problem

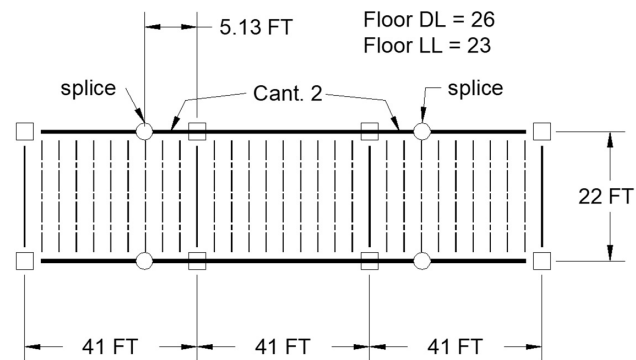
Given:

Span and loading

- $D + L = 49$ psf
- $49 \text{ psf} \times 11 \text{ ft} = 539 \text{ plf}$

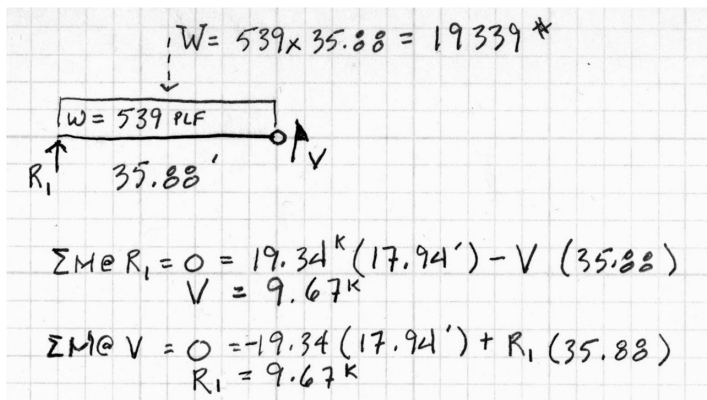
Find:

shear and moment
beam section



FBD 1

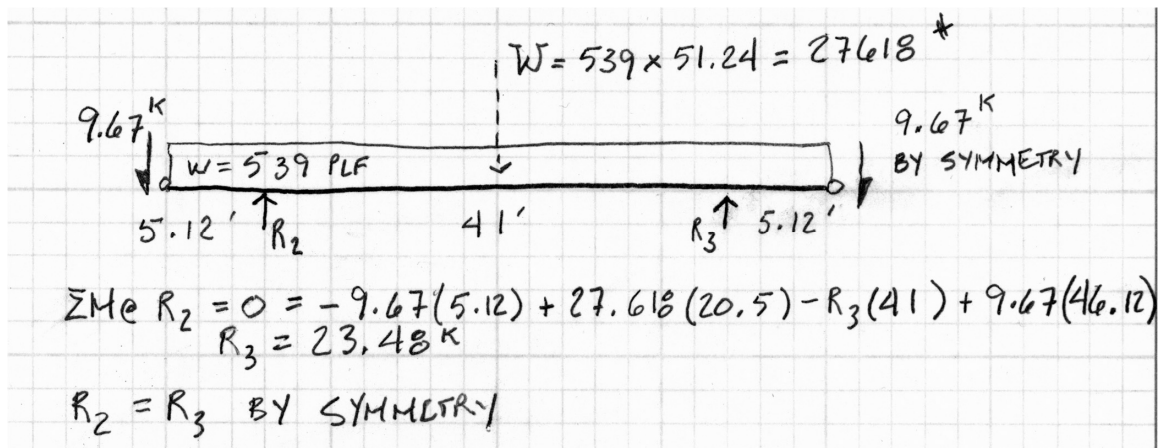
Reactions



Example Problem cont.

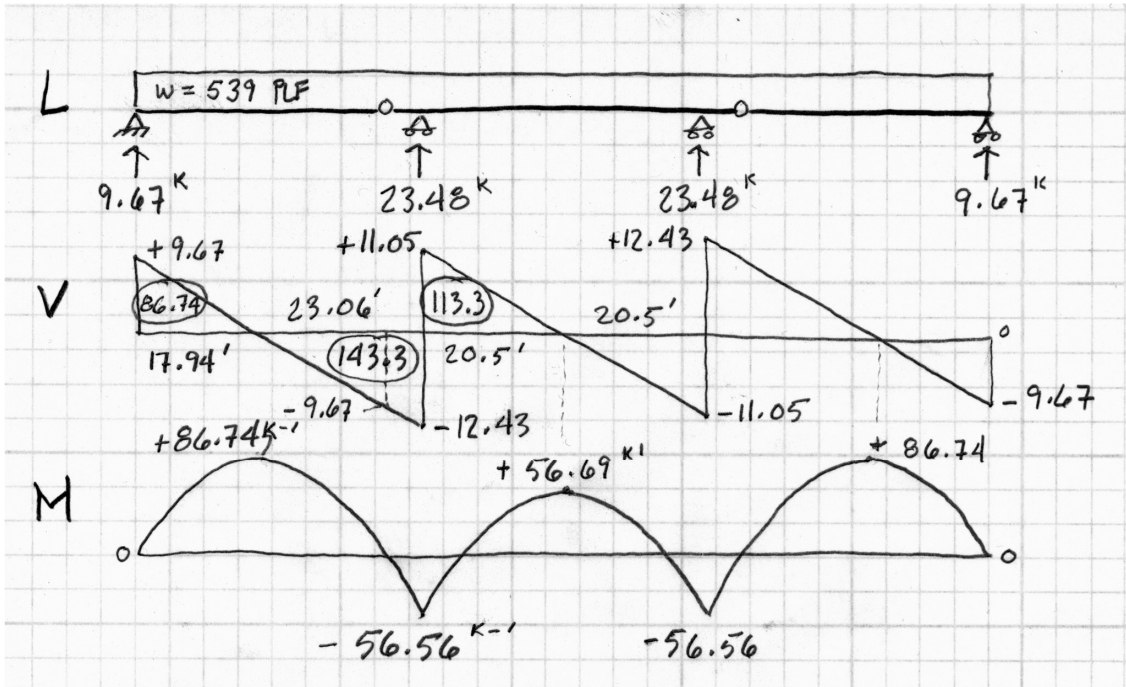
FBD 2

Reactions



Example Problem cont.

Force Diagrams



Example Problem cont.

STEEL BEAM DESIGN

$$M_U = 86.74 \text{ K-FT} \quad V_U = 12.43 \text{ K}$$

$$M_U = \phi M_n$$

$$M_n = \frac{M_U}{\phi} = \frac{86.74}{0.9} = 96.38 \text{ K-FT}$$

$$M_n = F_y Z_x$$

$$Z_x = \frac{M_n}{F_y} = \frac{96.38(12)}{50 \text{ KSI}} = 23.13 \text{ in}^3$$

CHECK SHEAR

$$h/t_w = 46.2 < 59 \quad \checkmark$$

$$A_w = t_w d = 0.235(12.5) = 2.87 \text{ in}^2$$

$$\phi V_n = (1.0) 0.6 F_y A_w = 0.6(50)(2.87)$$

$$\phi V_n = 86.01 > 12.43 = V_U \quad \checkmark \text{ OK}$$

Table 1-1 (continued)
W-Shapes
Dimensions

Shape	Area, A	Depth, d	Web			Flange			Distance					Workable Gage		
			Thickness, t _w	t _w / 2	Width, b _f	Thickness, t _f	k		k ₁	T						
							k _{end}	k _{det}								
	in. ²	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.			
W12x58	17.0	12.2	12 1/4	0.360	3/8	3/16	10.0	10	0.640	5/8	1.24	1 1/2	15/16	9/4	5 1/2	
x53	15.6	12.1	12	0.345	3/8	3/16	10.0	10	0.575	9/16	1.18	1 3/8	15/16	9/4	5 1/2	
W12x50	14.6	12.2	12 1/4	0.370	3/8	3/16	8.08	8 3/8	0.640	5/8	1.14	1 1/2	15/16	9/4	5 1/2	
x45	13.1	12.1	12	0.335	3/8	3/16	8.05	8	0.575	9/16	1.08	1 3/8	15/16	9/4	5 1/2	
x40	11.7	11.9	12	0.295	3/8	3/16	8.01	8	0.515	1/2	1.02	1 3/8	7/8			
W12x35	10.3	12.5	12 1/2	0.300	3/8	3/16	6.56	6 1/2	0.520	1/2	0.820	1 3/8	3/4	10 1/8	3 1/2	
x30	8.79	12.3	12 3/8	0.260	1/4	1/8	6.52	6 1/2	0.440	7/16	0.740	1 1/8	3/4			
x26	7.65	12.2	12 1/4	0.230	1/4	1/8	6.49	6 1/2	0.380	3/8	0.680	1 1/8	3/4			
W12x22	6.48	12.3	12 1/4	0.260	1/4	1/8	4.03	4	0.425	7/16	0.725	1 1/8	3/8	10 3/8	2 1/4	
x19	5.57	12.2	12 1/8	0.235	1/4	1/8	4.01	4	0.350	3/8	0.650	7/8	3/4			
x16	4.71	12.0	12	0.220	1/4	1/8	3.99	4	0.285	1/4	0.565	1 3/16	3/8			
x14	4.16	11.9	11 7/8	0.200	3/16	1/8	3.97	4	0.225	1/4	0.525	3/4	3/8			

Table 1-1 (continued)
W-Shapes
Properties

Nominal Wt. lb/ft	Compact Section Criteria		Axis X-X				Axis Y-Y				r_s	h_o	Torsional Properties	
	b_f/2t_f	h/t_w	I	S	r	Z	I	S	r	Z			J	C_w
58	7.82	27.0	475	78.0	5.28	86.4	107	21.4	2.51	32.5	2.81	11.6	2.10	3570
53	8.69	28.1	425	70.6	5.23	77.9	95.8	19.2	2.48	29.1	2.79	11.5	1.58	3160
50	6.31	26.8	391	64.2	5.18	71.9	56.3	13.9	1.96	21.3	2.25	11.6	1.71	1880
45	7.00	29.6	348	57.7	5.15	64.2	50.0	12.4	1.95	19.0	2.23	11.5	1.26	1650
40	7.77	33.6	307	51.5	5.13	57.0	44.1	11.0	1.94	16.8	2.21	11.4	0.906	1440
35	6.31	36.2	285	45.6	5.25	51.2	24.5	7.47	1.54	11.5	1.79	12.0	0.741	879
30	7.41	41.8	238	38.6	5.21	43.1	20.3	6.24	1.52	9.56	1.77	11.9	0.457	720
26	8.54	47.2	204	33.4	5.17	37.2	17.3	5.34	1.51	8.17	1.75	11.8	0.300	607
22	4.74	41.8	156	25.4	4.91	29.3	4.66	2.31	0.848	3.66	1.04	11.9	0.293	164
19	5.72	46.2	130	21.3	4.82	24.7	3.76	1.88	0.822	2.98	1.02	11.9	0.180	131
16	7.53	49.4	103	17.1	4.67	20.1	2.82	1.41	0.773	2.26	0.983	11.7	0.103	96.9
14	8.82	54.3	88.6	14.9	4.62	17.4	2.36	1.19	0.753	1.90	0.961	11.7	0.0704	80.4

Example Problem cont.

LOOK UP SECTION IN Z_x TABLE
 CHOOSE W12x19
 $Z_x = 24.7 > 23.13$ ✓
 $\phi M_n = 92.6 > 86.74$ ✓

Table 3-2 (continued)

W-Shapes

Selection by Z_x

Z_x

$F_y = 50$ ksi

Shape	Z_x	M_{px}/Ω_b		$\phi_b M_{px}$		M_{rx}/Ω_b		$\phi_b M_{rx}$		BF/Ω_b		$\phi_b BF$		L_p	L_r	I_x	V_{nx}/Ω_v		$\phi_v V_{nx}$	
		kip-ft	kip-ft	kip-ft	kip-ft	kip-ft	kip-ft	kip	kip	kip	kip	ASD	LRFD				ASD	LRFD		
	in. ³	ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD	ASD	LRFD	ft	ft			in. ⁴	ASD	LRFD	ASD	LRFD
W14x26	40.2	100	151	61.7	92.7	5.33	8.11	3.81	11.0	245	70.9	106								
W8x40	39.8	99.3	149	62.0	93.2	1.64	2.46	7.21	29.9	146	59.4	89.1								
W10x33	38.8	96.8	146	61.1	91.9	2.39	3.62	6.85	21.8	171	56.4	84.7								
W12x26	37.2	92.8	140	58.3	87.7	3.61	5.46	5.33	14.9	204	56.1	84.2								
W10x30	36.6	91.3	137	56.6	85.1	3.08	4.61	4.84	16.1	170	63.0	94.5								
W8x35	34.7	86.6	130	54.5	81.9	1.62	2.43	7.17	27.0	127	50.3	75.5								
W14x22	33.2	82.8	125	50.6	76.1	4.78	7.27	3.67	10.4	199	63.0	94.5								
W10x26	31.3	78.1	117	48.7	73.2	2.91	4.34	4.80	14.9	144	53.6	80.3								
W8x31	30.4	75.8	114	48.0	72.2	1.58	2.37	7.18	24.8	110	45.6	68.4								
W12x22	29.3	73.1	110	44.4	66.7	4.68	7.06	3.00	9.13	156	64.0	95.9								
W8x28	27.2	67.9	102	42.4	63.8	1.67	2.50	5.72	21.0	98.0	45.9	68.9								
W10x22	26.0	64.9	97.5	40.5	60.9	2.68	4.02	4.70	13.8	118	49.0	73.4								
W12x19	24.7	61.6	92.6	37.2	55.9	4.27	6.43	2.90	8.61	130	57.3	86.0								
W8x24	23.1	57.6	86.6	36.5	54.9	1.60	2.40	5.69	18.9	82.7	38.9	58.3								
W10x19	21.6	53.9	81.0	32.8	49.4	3.18	4.76	3.09	9.73	96.3	51.0	76.5								
W8x21	20.4	50.9	76.5	31.8	47.8	1.85	2.77	4.45	14.8	75.3	41.4	62.1								
W12x16	20.1	50.1	75.4	29.9	44.9	3.80	5.73	2.73	8.05	103	52.8	79.2								
W10x17	18.7	46.7	70.1	28.3	42.5	2.98	4.47	2.98	9.16	81.9	48.5	72.7								
W12x14	17.4	43.4	65.3	26.0	39.1	3.43	5.17	2.66	7.73	88.6	42.8	64.3								
W8x18	17.0	42.4	63.8	26.5	39.9	1.74	2.61	4.34	13.5	61.9	37.4	56.2								
W10x15	16.0	39.9	60.0	24.1	36.2	2.75	4.14	2.86	8.61	68.9	46.0	68.9								
W8x15	13.6	33.9	51.0	20.6	31.0	1.90	2.85	3.09	10.1	48.0	39.7	59.6								
W10x12	12.6	31.2	46.9	19.0	28.6	2.36	3.53	2.87	8.05	53.8	37.5	56.3								
W8x13	11.4	28.4	42.8	17.3	26.0	1.76	2.67	2.98	9.27	39.6	36.8	55.1								
W8x10	8.87	21.9	32.9	13.6	20.5	1.54	2.30	3.14	8.52	30.8	26.8	40.2								

ASD LRFD

$\Omega_b = 1.67$ $\phi_b = 0.90$
 $\Omega_v = 1.50$ $\phi_v = 1.00$

¹ Shape exceeds compact limit for flexure with $F_y = 50$ ksi; tabulated values have been adjusted accordingly.

² Shape does not meet the h/t_w limit for shear in AISC Specification Section G2.1(a) with $F_y = 50$ ksi; therefore, $\phi_v = 0.90$ and $\Omega_v = 1.67$.

Structural Optimization

Optimization procedure: Find the “best” solution for a given problem.

- Describe the goal – objectives (single vs. multiple)
- Determine limitations – constraints
- Describe the parameters – variables

Optimization type: What to optimize

- Material
- Member (section)
- Geometry
- Topology

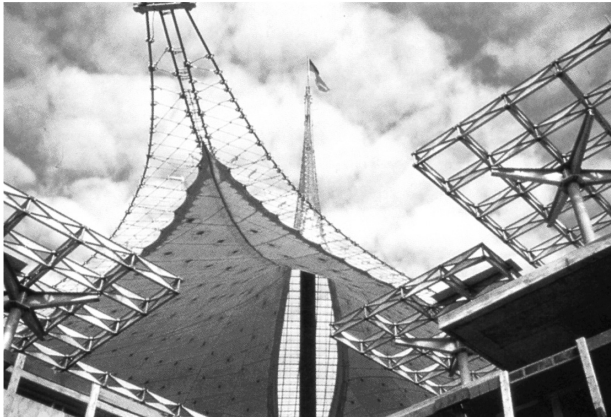


Optimization

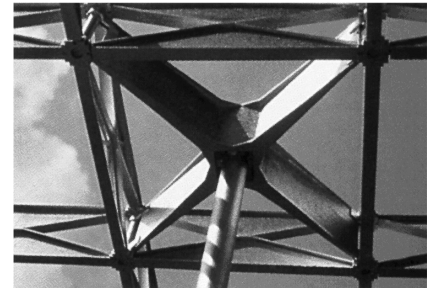
- Material
 - Composites
 - Steel vs. Aluminum
- Member and Geometry
 - Variable Depth or Width
 - Holes and Cut-outs



Biesenbach Viaduct, Blumberg Wutachtal Railroad, 1890
Eng. von Würthenau, Kräuter, Gebhard & Gernet



German Pavilion at Expo 1967, Montreal
Eng. Frei Otto Arch. Rolf Gutbrot



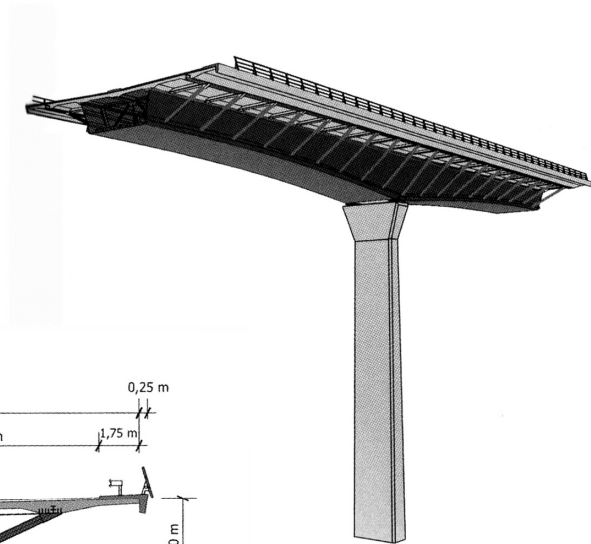
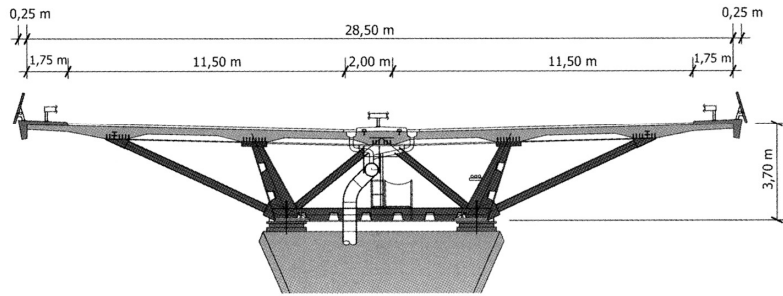
Section Optimization



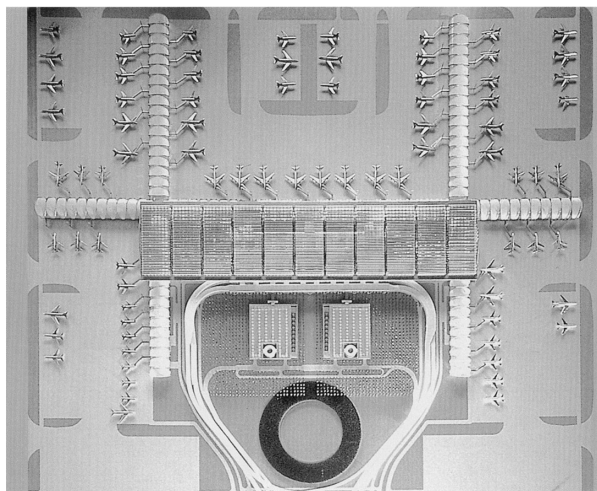
Reichenbach Valley Bridge, 2003
Eng. Büro Peter + Lochner

Section Optimization

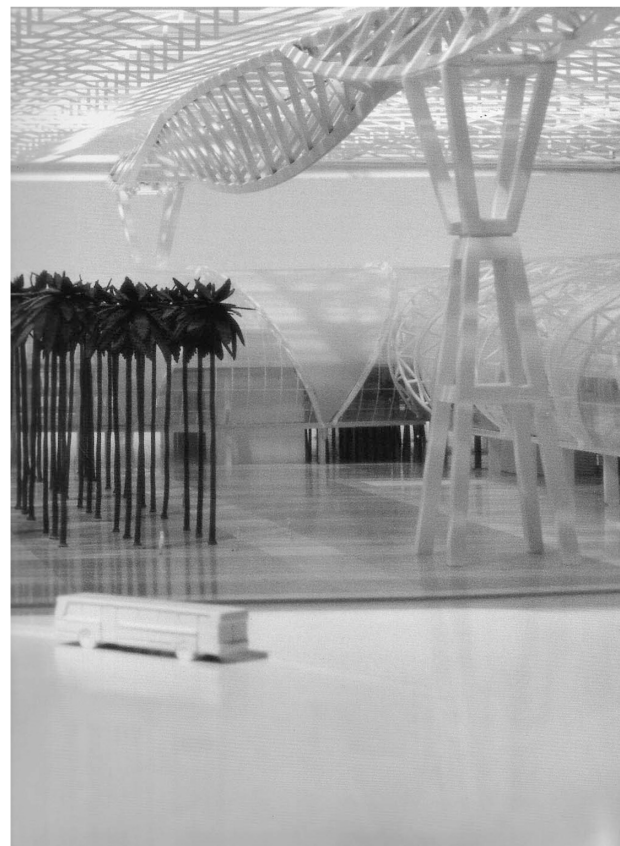
Reichenbach Valley Bridge, 2003
Eng. Büro Peter + Lochner



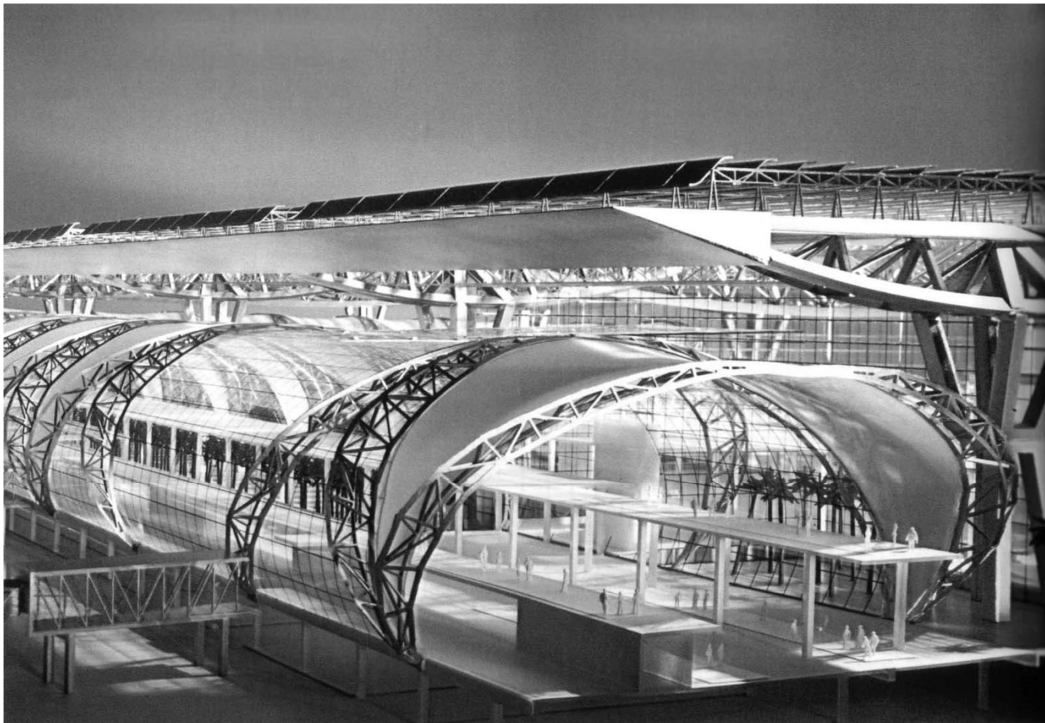
Geometry Optimization



New Bangkok International Airport, 2003
Eng. Werner Sobek Arch. Murphy Jahn

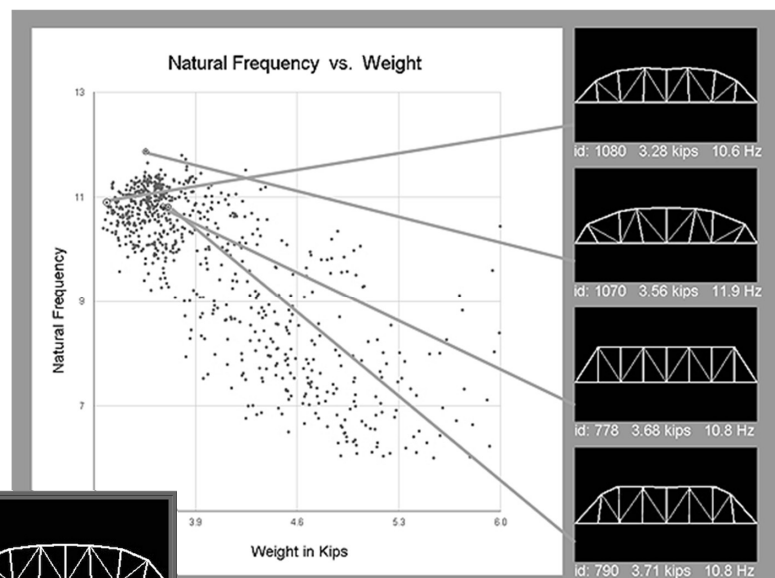
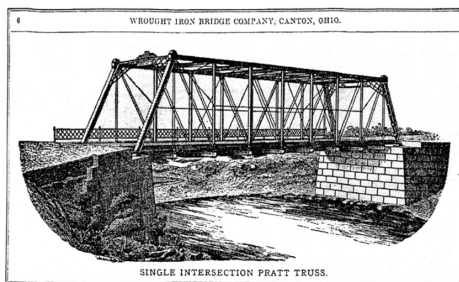


Geometry Optimization

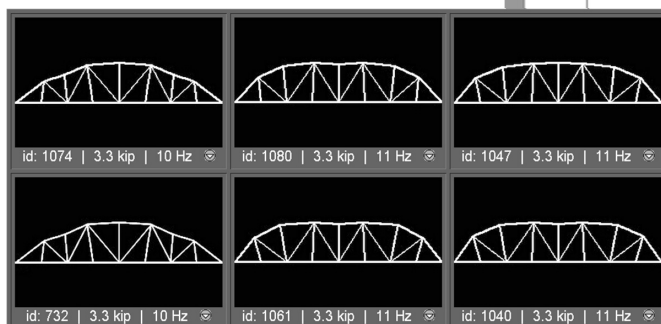


New Bangkok International Airport, 2003
Eng. Werner Sobek Arch. Murphy Jahn

Geometry Optimization - Bridges

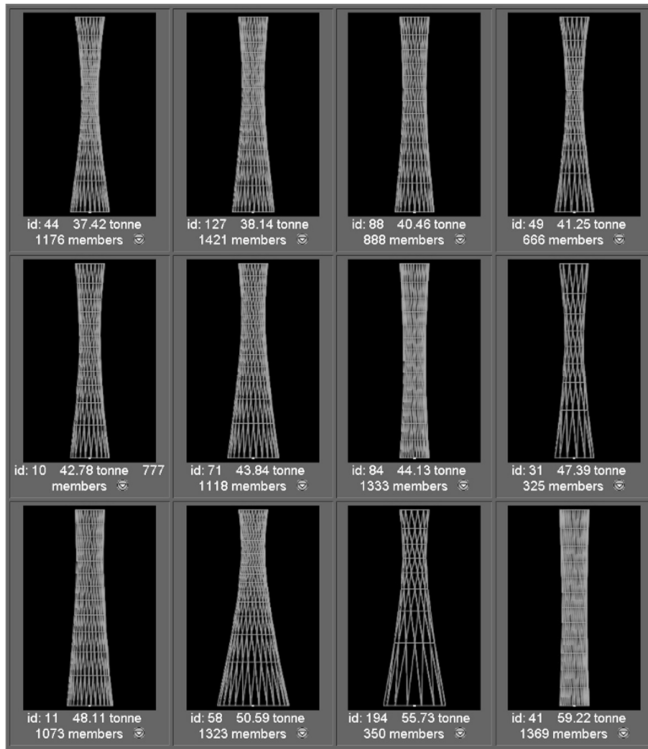


plot of weight vs natural frequency



lightest solutions

Topology Optimization - Shukhov towers



Nizhny Novgorod, 1896

