



ARCHITECTURE 324

Structures II

Recitation 07
Sections 04&05

Instructor
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GSI
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Feb 28, 2025

Office Hours

→ Office Hours

→ Day: Fridays, 12:00 PM - 1:00 PM

→ Location Options:

- In-person meetings: [2223B]
- Virtual meetings via Zoom

Please make sure to sign up at least 24 hours in advance to allow for proper scheduling via this link:

<https://docs.google.com/forms/d/e/1FAIpQLSdOb4gAc6SoCdsMAZP4zKrn3ecPyGt6dwVahVcOD3EqXGG-oA/viewform?usp=dialog>

If the slots are fully booked or if you have a time conflict, please email me directly to find an alternative time (arfazel@umich.edu)

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→ Continuous Beams

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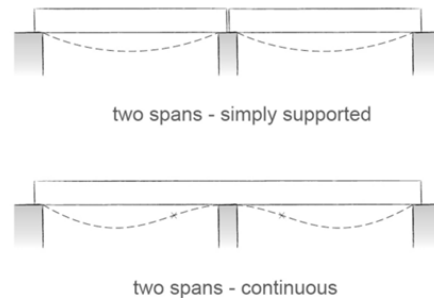
→ Lab

→ No lab for today

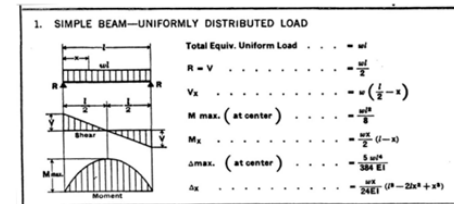
Structural Continuity

Continuous Beams

- Continuous over one or more supports
 - Most common in monolithic concrete
 - Steel: continuous or with moment connections
 - Wood: as continuous beams, affected long Glulam spans
- Statically indeterminate
 - Cannot be solved by the three equations of statics alone
 - Internal forces (shear & moment) as well as reactions are affected by movement or settlement of the supports

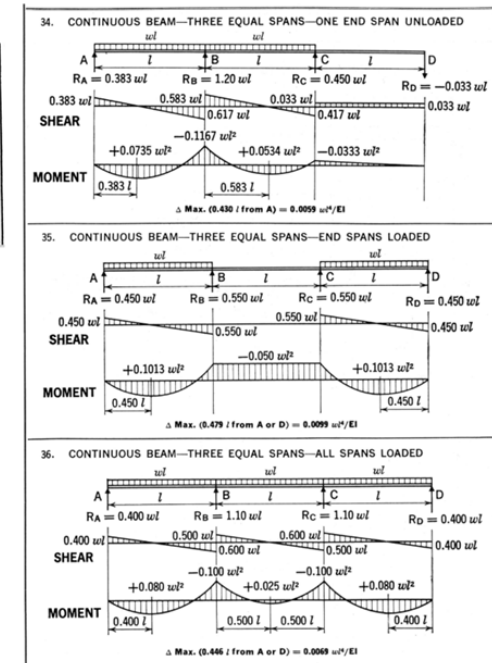


Simple vs. Continuous Beams



- Simple Beam
 - End moments = 0
 - $M_{\text{max at C.L.}} = wl^2/8 = 0.125wl^2$
- Continuous Beam
 - Exterior end moments = 0
 - Interior support moments are usually negative
 - Mid-span moments are usually positive

Note: moments shown reversed



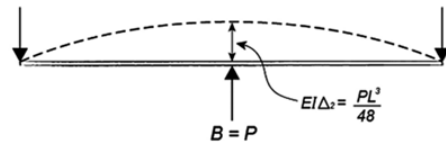
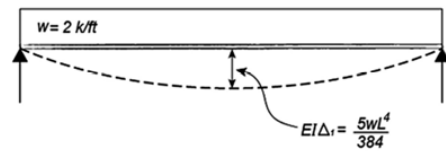
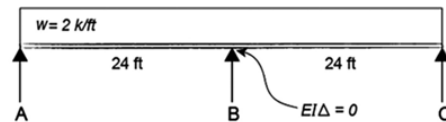
Structural Continuity

Deflection Method

- Two continuous, symmetric spans
- Symmetric Load

Procedure:

1. Remove the center support and calculate the center deflection for each load case as a simple span.
2. Remove the applied loads and replace the center support. Set the deflection equation for this case (center point load) equal to the deflection from step 1.
3. Solve the resulting equation for the center reaction force. (upward point load)
4. Calculate the remaining two end reactions.
5. Draw shear and moment diagrams as usual.

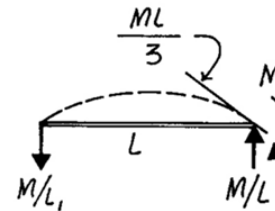


$$EI\Delta_1 + EI\Delta_2 = 0$$

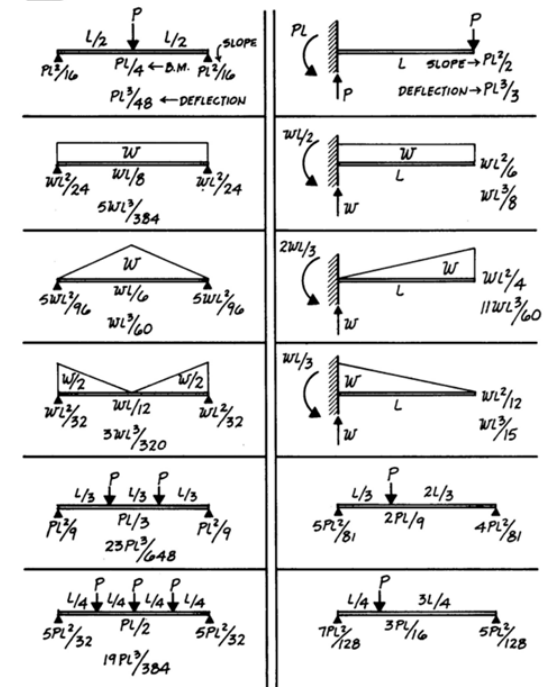
Slope Method

Slope equations:

$$M = \frac{3}{L_1 + L_2} [EI\Theta_1 + EI\Theta_2]$$

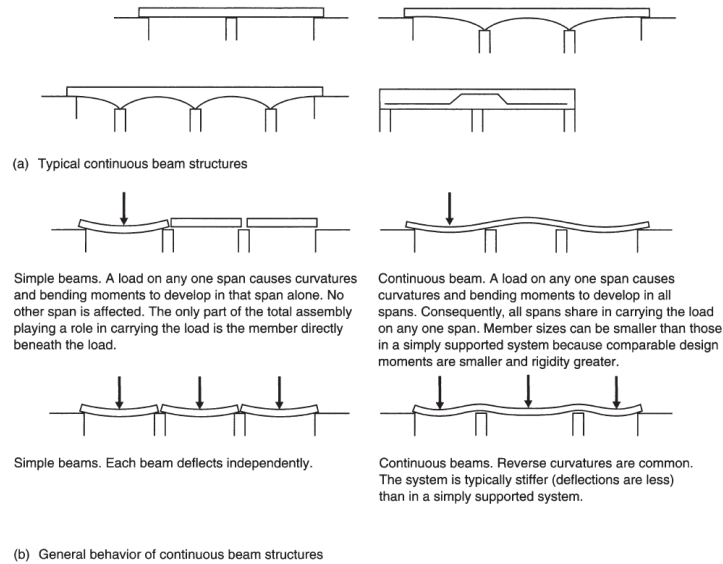


MAXIMUM VALUES: SLOPE, DEFLECTION, AND BENDING MOMENT
NOTE: VALUES OF SLOPE AND DEFLECTION TO BE DIVIDED BY "EI"



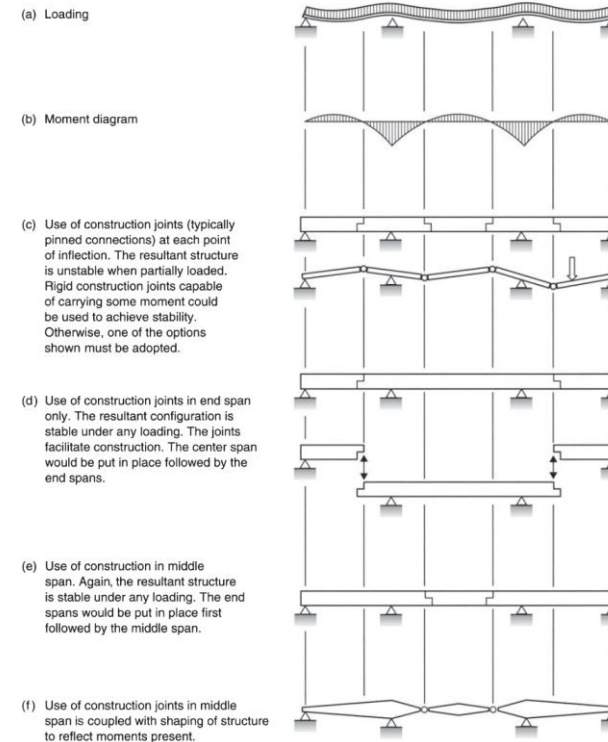
Structural Continuity

FIGURE 8.1 Continuous versus simple beams and typical types of continuous beams.



loads are often smaller than in comparable determinate structures. Thus, member sizes can often be reduced. Smaller amounts of material also are more efficiently used. Disadvantages in structures of this type include their sensitivity to support settlements and thermal effects. Support settlements, for example, can cause undesirable bending moments to develop in continuous beams over several supports, while not necessarily affecting a comparable series of simply supported beams.

FIGURE 8.14 Use of construction joints in continuous members. Construction joints often facilitate construction. Creating a condition of zero moment by design at points of inflection models the behavior of a continuous member by a series of statically determinate members.



Problem Set 06

6. Steel Column Analysis

For the given axially loaded steel W-section, determine the maximum floor live load capacity, P_{LL} . Assume the column is pinned top and bottom: $K = 1.0$, and there is no intermediate bracing. Use AISC-LRFD steel equations to determine ϕP_n and the load. $E = 29000$ ksi.

DATASET: 1

-2-

-3-

W-section

W8X31

F_y

36 KSI

Span A

32 FT

Span B

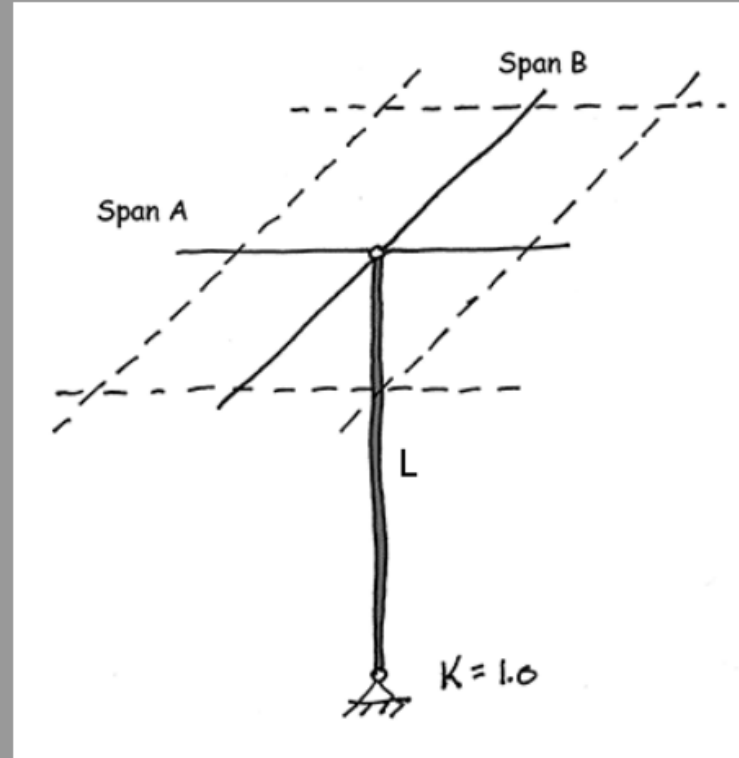
30 FT

Height L

17 FT

Floor Dead Load

39 PSF



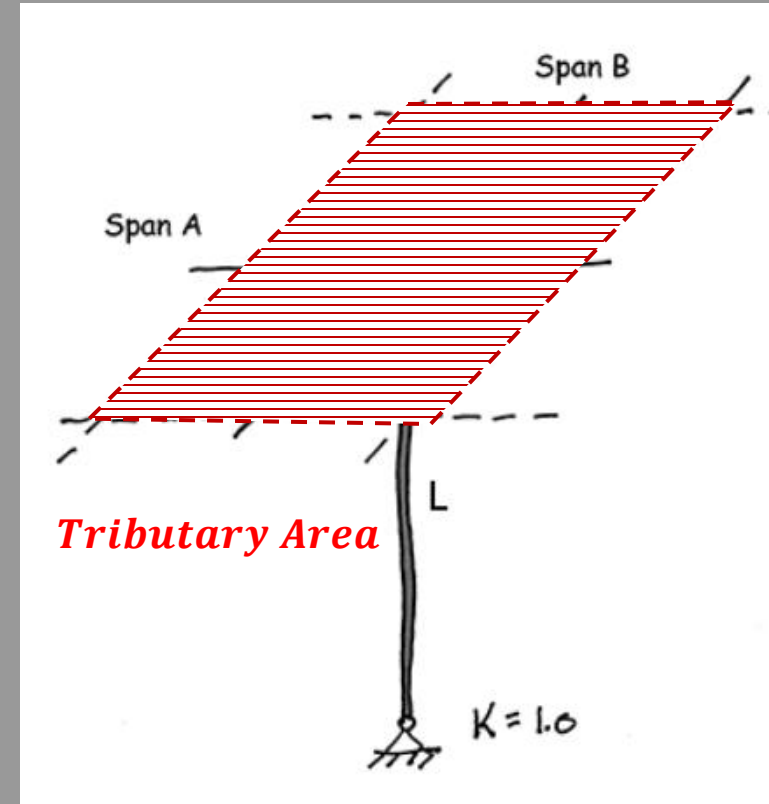
Problem Set 06

#Q1: Total unfactored floor dead load on the column

6. Steel Column Analysis

For the given axially loaded steel W-section, determine the maximum floor live load capacity, P_{LL} . Assume the column is pinned top and bottom: $K = 1.0$, and there is no intermediate bracing. Use AISC-LRFD steel equations to determine ϕP_n and the load. $E = 29000$ ksi.

DATASET: 1	-2-	-3-
W-section	W8X31	
Fy	36 KSI	
Span A	32 FT	
Span B	30 FT	
Height L	17 FT	
Floor Dead Load	39 PSF	



$$\text{Tributary Area} = \text{spanA} (\text{SpanB}) = 32(30) = 960 \text{ FT}^2$$

$$W_{DL} = \text{Floor}_{DL}(\text{Tributary Area}) = 39(960) = 37,440 \text{ lbs} = 37.44 \text{ KIPS}$$

Problem Set 06

#Q2: Controlling slenderness ratio

6. Steel Column Analysis

For the given axially loaded steel W-section, determine the maximum floor live load capacity, P_{LL}. Assume the column is pinned top and bottom: K = 1.0, and there is no intermediate bracing. Use AISC-LRFD steel equations to determine φ_pP_n and the load. E = 29000 ksi.

DATASET: 1 -2- -3-

W-section	W8X31
F _y	36 KSI
Span A	32 FT
Span B	30 FT
Height L	17 FT
Floor Dead Load	39 PSF

According to the table, for W8 × 31, we have:

$$r_x = 3.47 \text{ IN}, \quad r_y = 2.02 \text{ IN}$$

$$L_c = KL \quad (K = 1) \rightarrow L_c = L = 17 \text{ FT}$$

$$\text{slenderness ratio}_x = \frac{L_c}{r_x} = \frac{17 \text{ FT} \times \frac{12 \text{ IN}}{1 \text{ FT}}}{3.47} = 58.78$$

$$\text{slenderness ratio}_y = \frac{L_c}{r_y} = \frac{17 \times \frac{12 \text{ IN}}{1 \text{ FT}}}{2.02} = 100.99 > 58.78 \Rightarrow \text{we pick the bigger one}$$

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DIMENSIONS AND PROPERTIES


Table 1-1 (continued)
W-Shapes
Dimensions

Shape	Area, A	Depth, d	Web		Flange		Distance						Work- able Gage		
			Thickness, t _w	t _w / 2	Width, b _f	Thickness, t _f	k		k ₁	T					
	in. ²	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.	in.		
W8×67	19.7	9.00	9	0.570	9/16	5/16	8.28	8 1/4	0.935	13/16	1.33	1 1/8	19/16	5 3/4	5 1/2
×58	17.1	8.75	8 3/4	0.510	1/2	1/4	8.22	8 1/4	0.810	13/16	1.20	1 1/2	7/8		
×48	14.1	8.50	8 1/2	0.400	3/8	3/16	8.11	8 1/8	0.685	11/16	1.08	1 3/8	13/16		
×40	11.7	8.25	8 1/4	0.360	3/8	3/16	8.07	8 1/8	0.560	9/16	0.954	1 1/4	13/16		
×35	10.3	8.12	8 1/8	0.310	3/8	3/16	8.02	8	0.495	1/2	0.889	1 3/16	13/16		
×31 ^f	9.13	8.00	8	0.285	3/16	3/16	8.00	8	0.435	7/16	0.829	1 1/8	3/4		
W8×28	8.25	8.06	8	0.285	3/16	3/16	6.54	6 1/2	0.465	7/16	0.859	13/16	9/8	6 1/8	4
×24	7.08	7.93	7 7/8	0.245	1/4	1/8	6.50	6 1/2	0.400	3/8	0.794	7/8	9/16	6 1/8	4
W8×21	6.16	8.28	8 1/4	0.250	1/4	1/8	5.27	5 1/4	0.400	3/8	0.700	7/8	9/16	6 1/2	2 3/4 ^g
×18	5.26	8.14	8 1/8	0.230	1/4	1/8	5.25	5 1/4	0.330	9/16	0.630	13/16	9/16	6 1/2	2 3/4 ^g
W8×15	4.44	8.11	8 1/8	0.245	1/4	1/8	4.02	4	0.315	9/16	0.615	13/16	9/16	6 1/2	2 1/4 ^g
×13	3.84	7.99	8	0.230	1/4	1/8	4.00	4	0.255	1/4	0.555	3/4	9/16		
×10 ^{c,f}	2.96	7.89	7 7/8	0.170	3/16	1/8	3.94	4	0.205	3/16	0.505	11/16	1/2		
W6×25	7.34	6.38	6 3/8	0.320	5/16	3/16	6.08	6 1/8	0.455	7/16	0.705	15/16	9/16	4 1/2	3 1/2
×20	5.87	6.20	6 1/4	0.260	1/4	1/8	6.02	6	0.365	3/8	0.615	7/8	9/16		
×15 ^f	4.43	5.99	6	0.230	1/4	1/8	5.99	6	0.260	1/4	0.510	3/4	9/16		
W6×16	4.74	6.28	6 1/4	0.260	1/4	1/8	4.03	4	0.405	3/8	0.655	7/8	9/16	4 1/2	2 1/4 ^g
×12	3.55	6.03	6	0.230	1/4	1/8	4.00	4	0.280	1/4	0.530	3/4	9/16		
×9 ^f	2.68	5.90	5 7/8	0.170	3/16	1/8	3.94	4	0.215	3/16	0.465	11/16	1/2		
×8.5 ^f	2.52	5.83	5 7/8	0.170	3/16	1/8	3.94	4	0.195	3/16	0.445	11/16	1/2		
W5×19	5.56	5.15	5 1/8	0.270	1/4	1/8	5.03	5	0.430	7/16	0.730	13/16	7/16	3 1/2	2 3/4 ^g
×16	4.71	5.01	5	0.240	1/4	1/8	5.00	5	0.360	3/8	0.660	3/4	7/16	3 1/2	2 3/4 ^g
W4×13	3.83	4.16	4 1/8	0.280	1/4	1/8	4.06	4	0.345	3/8	0.595	3/4	1/2	2 5/8	2 1/4 ^g

DIMENSIONS AND PROPERTIES

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Table 1-1 (continued)
W-Shapes
Properties



Nom- inal Wt. lb/ft	Compact Section Criteria		Axis X-X				Axis Y-Y				r _{ts} in.	h _o in.	Torsional Properties	
	b _f /2t _f	h/t _w	I in. ⁴	S in. ³	r _x in.	Z in. ³	I in. ⁴	S in. ³	r _y in.	Z in. ³			J in. ⁴	C _w in. ⁶
67	4.43	11.1	272	60.4	3.72	70.1	88.6	21.4	2.12	32.7	2.43	8.07	5.05	1440
58	5.07	12.4	228	52.0	3.65	59.8	75.1	18.3	2.10	27.9	2.39	7.94	3.33	1180
48	5.92	15.9	184	43.2	3.61	49.0	60.9	15.0	2.08	22.9	2.35	7.82	1.96	931
40	7.21	17.6	146	35.5	3.53	39.8	49.1	12.2	2.04	18.5	2.31	7.69	1.12	726
35	8.10	20.5	127	31.2	3.51	34.7	42.6	10.6	2.03	16.1	2.28	7.63	0.769	619
31	9.19	22.3	110	27.5	3.47	30.4	37.1	9.27	2.02	14.1	2.26	7.57	0.536	530
28	7.03	22.3	98.0	24.3	3.45	27.2	21.7	6.63	1.82	10.1	1.84	7.60	0.537	312
24	8.12	25.9	82.7	20.9	3.42	23.1	18.3	5.63	1.61	8.57	1.81	7.53	0.346	259
21	6.59	27.5	75.3	18.2	3.49	20.4	9.77	3.71	1.26	5.69	1.46	7.88	0.282	152
18	7.95	29.9	61.9	15.2	3.43	17.0	7.97	3.04	1.23	4.66	1.43	7.81	0.172	122
15	6.37	28.1	48.0	11.8	3.29	13.6	3.41	1.70	0.876	2.67	1.06	7.80	0.137	51.8
13	7.84	29.9	39.6	9.91	3.21	11.4	2.73	1.37	0.843	2.15	1.03	7.74	0.0871	40.8
10	9.61	40.5	30.8	7.81	3.22	8.87	2.09	1.06	0.841	1.66	1.01	7.69	0.0426	30.9
25	6.68	15.5	53.4	16.7	2.70	18.9	17.1	5.61	1.52	8.56	1.74	5.93	0.461	150
20	8.25	19.1	41.4	13.4	2.66	14.9	13.3	4.41	1.50	6.72	1.70	5.84	0.240	113
15	11.5	21.6	29.1	9.72	2.56	10.8	9.32	3.11	1.45	4.75	1.66	5.73	0.101	76.5
16	4.98	19.1	32.1	10.2	2.60	11.7	4.43	2.20	0.967	3.39	1.13	5.88	0.223	38.2
12	7.14	21.6	22.1	7.31	2.49	8.30	2.99	1.50	0.918	2.32	1.08	5.75	0.0903	24.7
9	9.16	29.2	16.4	5.56	2.47	6.23	2.20	1.11	0.905	1.72	1.06	5.69	0.0405	17.7
8.5	10.1	29.1	14.9	5.10	2.43	5.73	1.99	1.01	0.890	1.56	1.05	5.64	0.0333	15.8
19	5.85	13.7	26.3	10.2	2.17	11.6	9.13	3.63	1.28	5.53	1.45	4.72	0.316	50.9
16	6.94	15.4	21.4	8.55	2.13	9.63	7.51	3.00	1.26	4.58	1.43	4.65	0.192	40.6
13	5.88	10.6	11.3	5.46	1.72	6.28	3.86	1.90	1.00	2.92	1.16	3.82	0.151	14.0

Problem Set 06

#Q3: Transition slenderness value, $4.71(E/F_y)^{.5}$

6. Steel Column Analysis

For the given axially loaded steel W-section, determine the maximum floor live load capacity, P_{LL}. Assume the column is pinned top and bottom: K = 1.0, and there is no intermediate bracing. Use AISC-LRFD steel equations to determine ϕP_n and the load. E = 29000 ksi.

DATASET: 1	-2-	-3-
W-section	W8X31	
F _y	36 KSI	
Span A	32 FT	
Span B	30 FT	
Height L	17 FT	
Floor Dead Load	39 PSF	

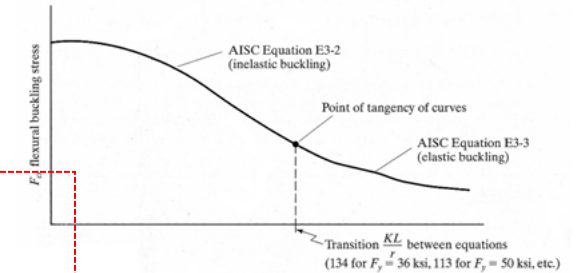
$$\text{Transition slenderness} = 4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{29000}{36}} = 134$$
$$100.99 < 134 \Rightarrow \text{short}$$

Analysis of Steel Columns

Short columns
Fail by material crushing
Plastic behavior

Intermediate columns
Crush partially and then buckle
Inelastic behavior
Local buckling – flange or web
Flexural torsional buckling - twisting

Long columns
Fail in Euler buckling
Elastic behavior



$$\text{slenderness} = \frac{KL}{r}$$

short intermediate long

Transition Slenderness $4.71 \sqrt{\frac{E}{F_y}}$

Problem Set 06

#Q4: Euler stress, F_e

#Q5: Critical stress, F_{cr}

DATASET: 1

-2-

-3-

W-section	W8X31
F_y	36 KSI
Span A	32 FT
Span B	30 FT
Height L	17 FT
Floor Dead Load	39 PSF

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = \frac{3.14^2 (29000)}{(100.99)^2} = \mathbf{28.06 \text{ KSI}}$$

$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}} \right] F_y = \left[0.658^{\frac{36}{28.06}} \right] (36) = \mathbf{21.04 \text{ KSI}}$$

Analysis of Steel Columns - LRFD

Euler equation:

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2}$$

Short & Intermediate Columns:

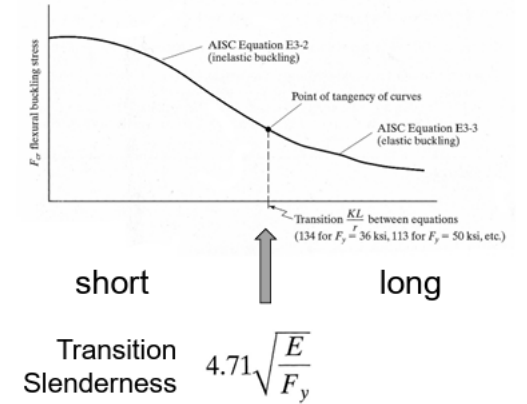
$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}} \right] F_y$$

Equation E3-2

Long Columns:

$$F_{cr} = 0.877 F_e$$

Equation E3-3



$$P_n = F_{cr} A_g$$

$$\phi_c P_n = \phi_c F_{cr} A_g$$

$$(\phi_c = 0.90)$$

Problem Set 06

#Q6: Nominal strength, P_n

#Q7: Factored nominal strength, ϕP_n

DATASET: 1

-2-

-3-

W-section

W8X31

F_y

36 KSI

Span A

32 FT

Span B

30 FT

Height L

17 FT

Floor Dead Load

39 PSF

$$A_g = 9.13 \text{ IN}^2$$

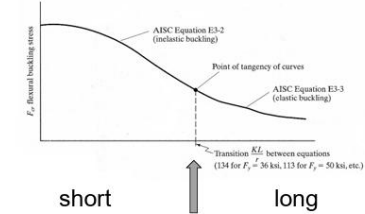
$$P_n = F_{cr} A_g = 21.04 (9.13) = 192.09 \text{ KIPS}$$

$$\phi P_n = 192.09 (0.9) = 172.88 \text{ KIPS}$$

Analysis of Steel Columns - LRFD

Euler equation:

$$F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2}$$



Short & Intermediate Columns:

$$F_{cr} = \left[0.658^{\frac{F_y}{F_e}} \right] F_y$$

Equation E3-2

Long Columns:

$$F_{cr} = 0.877 F_e$$

Equation E3-3

$$\text{Transition Slenderness} = 4.71 \sqrt{\frac{E}{F_y}}$$

$$P_n = F_{cr} A_g$$

$$\phi_c P_n = \phi_c F_{cr} A_g$$

$$(\phi_c = 0.90)$$

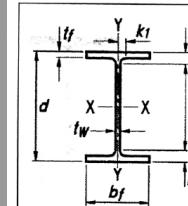


Table 1-1 (continued)
W-Shapes
Dimensions

Shape	Area, A	Depth, d	Web			Flange			Distance					
			Thickness, t_w	$\frac{t_w}{2}$		Width, b_f	Thickness, t_f	k		k_1	T	Workable Gage		
	in. ²		in.	in.	in.	in.	in.	k_{des}	k_{det}	in.	in.	in.		
W8×67	19.7	9.00	9	0.570	$\frac{9}{16}$	$\frac{5}{16}$	8.28	$8\frac{1}{4}$	0.935	$\frac{15}{16}$	1.33	$\frac{15}{8}$	$\frac{15}{16}$	$5\frac{1}{2}$
×58	17.1	8.75	$8\frac{3}{4}$	0.510	$\frac{1}{2}$	$\frac{1}{4}$	8.22	$8\frac{1}{4}$	0.810	$\frac{13}{16}$	1.20	$1\frac{1}{2}$	$\frac{7}{8}$	$5\frac{1}{2}$
×48	14.1	8.50	$8\frac{1}{2}$	0.400	$\frac{3}{8}$	$\frac{3}{16}$	8.11	$8\frac{1}{8}$	0.685	$\frac{11}{16}$	1.08	$\frac{13}{8}$	$\frac{13}{16}$	$5\frac{1}{2}$
×40	11.7	8.25	$8\frac{1}{4}$	0.360	$\frac{3}{8}$	$\frac{3}{16}$	8.07	$8\frac{1}{8}$	0.560	$\frac{9}{16}$	0.954	$1\frac{1}{4}$	$\frac{13}{16}$	$5\frac{1}{2}$
×35	10.3	8.12	$8\frac{1}{8}$	0.310	$\frac{5}{16}$	$\frac{3}{16}$	8.02	8	0.495	$\frac{1}{2}$	0.889	$\frac{13}{16}$	$\frac{13}{16}$	$5\frac{1}{2}$
×31 ¹	9.13	8.00	8	0.285	$\frac{5}{16}$	$\frac{3}{16}$	8.00	8	0.435	$\frac{7}{16}$	0.829	$\frac{1}{8}$	$\frac{3}{4}$	$5\frac{1}{2}$
W8×28	8.25	8.06	8	0.285	$\frac{5}{16}$	$\frac{1}{16}$	6.54	$6\frac{1}{2}$	0.465	$\frac{7}{16}$	0.858	$\frac{15}{16}$	$\frac{5}{8}$	4
×24	7.08	7.93	$7\frac{7}{8}$	0.245	$\frac{1}{4}$	$\frac{3}{16}$	6.50	$6\frac{1}{2}$	0.400	$\frac{3}{8}$	0.794	$\frac{7}{8}$	$\frac{9}{16}$	4
W8×21	6.16	8.28	$8\frac{1}{4}$	0.250	$\frac{1}{4}$	$\frac{1}{8}$	5.27	$5\frac{1}{4}$	0.400	$\frac{3}{8}$	0.700	$\frac{7}{8}$	$\frac{9}{16}$	$2\frac{3}{4}$ ²

Problem Set 06

#Q8: UN-factored live load on column (actual total LL)

#Q9: Actual unfactored floor live load

DATASET: 1	-2-	-3-
W-section	W8X31	
Fy	36 KSI	
Span A	32 FT	
Span B	30 FT	
Height L	17 FT	
Floor Dead Load	39 PSF	

$$P_u = 1.2 (DL) + 1.6 (LL), \text{ (max load)} \rightarrow P_u = \phi P_n$$

$$172.88 = 1.2 (37.44) + 1.6 (LL)$$

$$\Rightarrow LL = 79.97 \text{ KIPS}$$

$$\Rightarrow \text{floor live load} = \frac{LL}{\text{Tributary Area}} = \frac{79.97 * 1000}{960} = 83.30 \text{ PSF}$$

Procedure - Analysis of Steel Beams – for Zone 1 $L_b < L_p$

Pass/Fail

Given: yield stress, steel section, loading, bracing (L_b)

Find: pass/fail of section

1. Calculate the factored design load w_u
 $w_u = 1.2w_{DL} + 1.6w_{LL}$

2. Determine the design moment M_u .
 M_u will be the maximum beam moment using the factored loads

3. Insure that $L_b < L_p$ (zone 1)
 $L_p = 1.76 r_y \sqrt{E/F_y}$

4. Determine the nominal moment, M_n
 $M_n = F_y Z_x$ (look up Z_x for section)

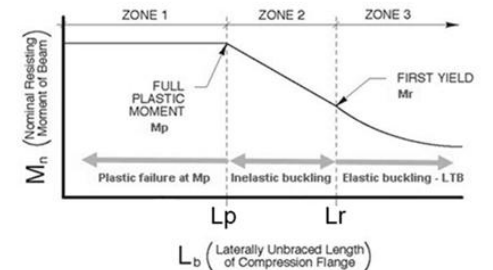
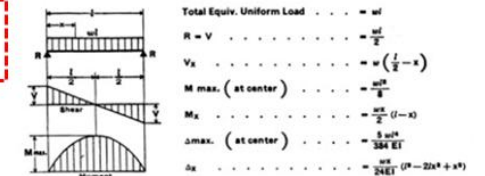
5. Factor the nominal moment
 $\phi M_n = 0.90 M_n$

6. Check that $M_u < \phi M_n$

7. Check shear

8. Check deflection

1. SIMPLE BEAM—UNIFORMLY DISTRIBUTED LOAD



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Structures II

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No lab for today ;)